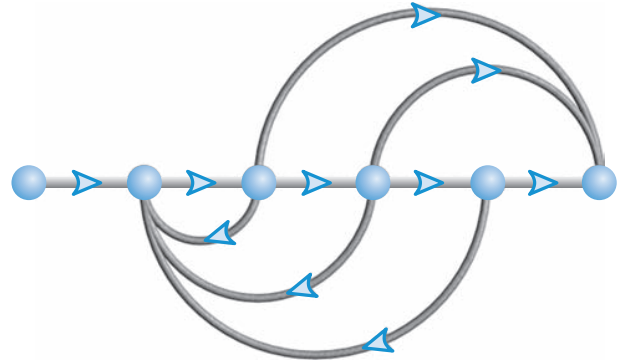


# Modeling in the Time Domain

# 3



This chapter covers only state-space methods.

State Space  
SS

## Chapter Learning Outcomes

After completing this chapter, the student will be able to:

- Find a mathematical model, called a *state-space* representation, for a linear, time-invariant system (Sections 3.1–3.3)
- Model electrical and mechanical systems in state space (Section 3.4)
- Convert a transfer function to state space (Section 3.5)
- Convert a state-space representation to a transfer function (Section 3.6)
- Linearize a state-space representation (Section 3.7)

## Case Study Learning Outcomes

You will be able to demonstrate your knowledge of the chapter objectives with case studies as follows:

- Given the antenna azimuth position control system shown on the front endpapers, you will be able to find the state-space representation of each subsystem.
- Given a description of the way a pharmaceutical drug flows through a human being, you will be able to find the state-space representation to determine drug concentrations in specified compartmentalized blocks of the process and of the human body. You will also be able to apply the same concepts to an aquifer to find water level.

## 3.1 Introduction

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Two approaches are available for the analysis and design of feedback control systems. The first, which we began to study in Chapter 2, is known as the *classical*, or *frequency-domain*, technique. This approach is based on converting a system's differential equation to a transfer function, thus generating a mathematical model of the system that *algebraically* relates a representation of the output to a representation of the input. Replacing a differential equation with an algebraic equation not only simplifies the representation of individual subsystems but also simplifies modeling interconnected subsystems.

The primary disadvantage of the classical approach is its limited applicability: It can be applied only to linear, time-invariant systems or systems that can be approximated as such.

A major advantage of frequency-domain techniques is that they rapidly provide stability and transient response information. Thus, we can immediately see the effects of varying system parameters until an acceptable design is met.

With the arrival of space exploration, requirements for control systems increased in scope. Modeling systems by using linear, time-invariant differential equations and subsequent transfer functions became inadequate. The *state-space* approach (also referred to as the *modern*, or *time-domain*, approach) is a unified method for modeling, analyzing, and designing a wide range of systems. For example, the state-space approach can be used to represent nonlinear systems that have backlash, saturation, and dead zone. Also, it can handle, conveniently, systems with nonzero initial conditions. Time-varying systems, (for example, missiles with varying fuel levels or lift in an aircraft flying through a wide range of altitudes) can be represented in state space. Many systems do not have just a single input and a single output. Multiple-input, multiple-output systems (such as a vehicle with input direction and input velocity yielding an output direction and an output velocity) can be compactly represented in state space with a model similar in form and complexity to that used for single-input, single-output systems. The time-domain approach can be used to represent systems with a digital computer in the loop or to model systems for digital simulation. With a simulated system, system response can be obtained for changes in system parameters—an important design tool. The state-space approach is also attractive because of the availability of numerous state-space software packages for the personal computer.

The time-domain approach can also be used for the same class of systems modeled by the classical approach. This alternate model gives the control systems designer another perspective from which to create a design. While the state-space approach can be applied to a wide range of systems, it is not as intuitive as the classical approach. The designer has to engage in several calculations before the physical interpretation of the model is apparent, whereas in classical control a few quick calculations or a graphic presentation of data rapidly yields the physical interpretation.

In this book, the coverage of state-space techniques is to be regarded as an introduction to the subject, a springboard to advanced studies, and an alternate approach to frequency-domain techniques. We will limit the state-space approach to linear, time-invariant systems or systems that can be linearized by the methods of Chapter 2. The study of other classes of systems is beyond the scope of this book. Since state-space analysis and design rely on matrices and matrix operations, you may want to review this topic in Appendix G, located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise), before continuing.

## 3.2 Some Observations

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We proceed now to establish the state-space approach as an alternate method for representing physical systems. This section sets the stage for the formal definition of the state-space representation by making some observations about systems and their variables. In the discussion that follows, some of the development has been placed in

footnotes to avoid clouding the main issues with an excess of equations and to ensure that the concept is clear. Although we use two electrical networks to illustrate the concepts, we could just as easily have used a mechanical or any other physical system.

We now demonstrate that for a system with many variables, such as inductor voltage, resistor voltage, and capacitor charge, we need to use differential equations only to solve for a selected subset of system variables because all other remaining system variables can be evaluated algebraically from the variables in the subset. Our examples take the following approach:

1. We select a particular *subset* of all possible system variables and call the variables in this subset *state variables*.
2. For an  $n$ th-order system, we write  $n$  *simultaneous, first-order differential equations* in terms of the state variables. We call this system of simultaneous differential equations *state equations*.
3. If we know the initial condition of all of the state variables at  $t_0$  as well as the system input for  $t \geq t_0$ , we can solve the simultaneous differential equations for the state variables for  $t \geq t_0$ .
4. We *algebraically* combine the state variables with the system's input and find all of the other system variables for  $t \geq t_0$ . We call this algebraic equation the *output equation*.
5. We consider the state equations and the output equations a viable representation of the system. We call this representation of the system a *state-space representation*.

Let us now follow these steps through an example. Consider the  $RL$  network shown in Figure 3.1 with an initial current of  $i(0)$ .

1. We select the current,  $i(t)$ , for which we will write and solve a differential equation using Laplace transforms.
2. We write the loop equation,

$$L \frac{di}{dt} + Ri = v(t) \quad (3.1)$$

3. Taking the Laplace transform, using Table 2.2, Item 7, and including the initial conditions, yields

$$L[sI(s) - i(0)] + RI(s) = V(s) \quad (3.2)$$

Assuming the input,  $v(t)$ , to be a unit step,  $u(t)$ , whose Laplace transform is  $V(s) = 1/s$ , we solve for  $I(s)$  and get

$$I(s) = \frac{1}{R} \left( \frac{1}{s} - \frac{1}{s + \frac{R}{L}} \right) + \frac{i(0)}{s + \frac{R}{L}} \quad (3.3)$$

from which

$$i(t) = \frac{1}{R} \left( 1 - e^{-(R/L)t} \right) + i(0)e^{-(R/L)t} \quad (3.4)$$

The function  $i(t)$  is a subset of all possible network variables that we are able to find from Eq. (3.4) if we know its initial condition,  $i(0)$ , and the input,  $v(t)$ . Thus,  $i(t)$  is a state variable, and the differential equation (3.1) is a *state equation*.

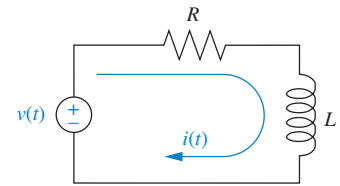


FIGURE 3.1  $RL$  network

4. We can now solve for all of the other network variables *algebraically* in terms of  $i(t)$  and the applied voltage,  $v(t)$ . For example, the voltage across the resistor is

$$v_R(t) = Ri(t) \quad (3.5)$$

The voltage across the inductor is

$$v_L(t) = v(t) - Ri(t) \quad (3.6)^1$$

The derivative of the current is

$$\frac{di}{dt} = \frac{1}{L}[v(t) - Ri(t)] \quad (3.7)^2$$

Thus, knowing the state variable,  $i(t)$ , and the input,  $v(t)$ , we can find the value, or *state*, of any network variable at any time,  $t \geq t_0$ . Hence, the algebraic equations, Eqs. (3.5) through (3.7), are *output equations*.

5. Since the variables of interest are completely described by Eq. (3.1) and Eqs. (3.5) through (3.7), we say that the combined state equation (3.1) and the output equations (3.5 through 3.7) form a viable representation of the network, which we call a *state-space representation*.

Equation (3.1), which describes the dynamics of the network, is not unique. This equation could be written in terms of any other network variable. For example, substituting  $i = v_R/R$  into Eq. (3.1) yields

$$\frac{L}{R} \frac{dv_R}{dt} + v_R = v(t) \quad (3.8)$$

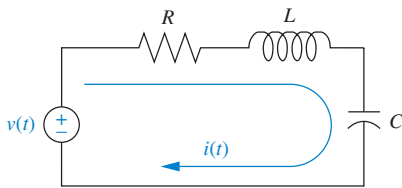


FIGURE 3.2 RLC network

which can be solved knowing that the initial condition  $v_R(0) = Ri(0)$  and knowing  $v(t)$ . In this case, the state variable is  $v_R(t)$ . Similarly, all other network variables can now be written in terms of the state variable,  $v_R(t)$ , and the input,  $v(t)$ . Let us now extend our observations to a second-order system, such as that shown in Figure 3.2.

1. Since the network is of second order, two simultaneous, first-order differential equations are needed to solve for two state variables. We select  $i(t)$  and  $q(t)$ , the charge on the capacitor, as the two state variables.
2. Writing the loop equation yields

$$L \frac{di}{dt} + Ri + \frac{1}{C} \int i dt = v(t) \quad (3.9)$$

Converting to charge, using  $i(t) = dq/dt$ , we get

$$L \frac{d^2q}{dt^2} + R \frac{dq}{dt} + \frac{1}{C} q = v(t) \quad (3.10)$$

<sup>1</sup> Since  $v_L(t) = v(t) - v_R(t) = v(t) - Ri(t)$ .

<sup>2</sup> Since  $\frac{di}{dt} = \frac{1}{L} v_L(t) = \frac{1}{L} [v(t) - Ri(t)]$ .

But an  $n$ th-order differential equation can be converted to  $n$  simultaneous first-order differential equations, with each equation of the form

$$\frac{dx_i}{dt} = a_{i1}x_1 + a_{i2}x_2 + \cdots + a_{in}x_n + b_i f(t) \quad (3.11)$$

where each  $x_i$  is a state variable, and the  $a_{ij}$ 's and  $b_i$  are constants for linear, time-invariant systems. We say that the right-hand side of Eq. (3.11) is a *linear combination* of the state variables and the input,  $f(t)$ .

We can convert Eq. (3.10) into two simultaneous, first-order differential equations in terms of  $i(t)$  and  $q(t)$ . The first equation can be  $dq/dt = i$ . The second equation can be formed by substituting  $\int i dt = q$  into Eq. (3.9) and solving for  $di/dt$ . Summarizing the two resulting equations, we get

$$\frac{dq}{dt} = i \quad (3.12a)$$

$$\frac{di}{dt} = -\frac{1}{LC}q - \frac{R}{L}i + \frac{1}{L}v(t) \quad (3.12b)$$

3. These equations are the state equations and can be solved simultaneously for the state variables,  $q(t)$  and  $i(t)$ , using the Laplace transform and the methods of Chapter 2. In addition we must also know the input,  $v(t)$ , and the initial conditions for  $q(t)$  and  $i(t)$ .
4. From these two state variables, we can solve for all other network variables. For example, the voltage across the inductor can be written in terms of the solved state variables and the input as

$$v_L(t) = -\frac{1}{C}q(t) - Ri(t) + v(t) \quad (3.13)^3$$

Equation (3.13) is an *output equation*; we say that  $v_L(t)$  is a *linear combination* of the state variables,  $q(t)$  and  $i(t)$ , and the input,  $v(t)$ .

5. The combined state equations (3.12) and the output equation (3.13) form a viable representation of the network, which we call a *state-space representation*.

Another choice of two state variables can be made, for example,  $v_R(t)$  and  $v_C(t)$ , the resistor and capacitor voltage, respectively. The resulting set of simultaneous, first-order differential equations follows:

$$\frac{dv_R}{dt} = -\frac{R}{L}v_R - \frac{R}{L}v_C + \frac{R}{L}v(t) \quad (3.14a)^4$$

$$\frac{dv_C}{dt} = \frac{1}{RC}v_R \quad (3.14b)$$

Again, these differential equations can be solved for the state variables if we know the initial conditions along with  $v(t)$ . Further, all other network variables can be found as a linear combination of these state variables.

Is there a restriction on the choice of state variables? Yes! Typically, the minimum number of state variables required to describe a system equals the order of the differential equation. Thus, a second-order system requires a minimum of two state variables to describe it.

<sup>3</sup> Since  $v_L(t) = L(di/dt) = -(1/C)q - Ri + v(t)$ , where  $di/dt$  can be found from Eq. (3.9), and  $\int i dt = q$ .

<sup>4</sup> Since  $v_R(t) = i(t)R$ , and  $v_C(t) = (1/C)\int i dt$ , differentiating  $v_R(t)$  yields  $dv_R/dt = R(di/dt) = (R/L)v_L = (R/L)[v(t) - v_R - v_C]$ , and differentiating  $v_C(t)$  yields  $dv_C/dt = (1/C)i = (1/RC)v_R$ .

We can define more state variables than the minimal set; however, within this minimal set the state variables must be linearly independent. For example, if  $v_R(t)$  is chosen as a state variable, then  $i(t)$  cannot be chosen, because  $v_R(t)$  can be written as a linear combination of  $i(t)$ , namely  $v_R(t) = Ri(t)$ . Under these circumstances we say that the state variables are *linearly dependent*. State variables must be *linearly independent*; that is, no state variable can be written as a linear combination of the other state variables, or else we would not have enough information to solve for all other system variables, and we could even have trouble writing the simultaneous equations themselves.

The state and output equations can be written in vector-matrix form if the system is linear. Thus, Eq. (3.12), the state equations, can be written as

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u \quad (3.15)$$

where

$$\dot{\mathbf{x}} = \begin{bmatrix} dq/dt \\ di/dt \end{bmatrix}; \quad \mathbf{A} = \begin{bmatrix} 0 & 1 \\ -1/LC & -R/L \end{bmatrix}$$

$$\mathbf{x} = \begin{bmatrix} q \\ i \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} 0 \\ 1/L \end{bmatrix}; \quad u = v(t)$$

Equation (3.13), the output equation, can be written as

$$y = \mathbf{C}\mathbf{x} + Du \quad (3.16)$$

where

$$y = v_L(t); \quad \mathbf{C} = [-1/C \quad -R]; \quad \mathbf{x} = \begin{bmatrix} q \\ i \end{bmatrix}; \quad D = 1; \quad u = v(t)$$

We call the combination of Eqs. (3.15) and (3.16) a *state-space representation* of the network of Figure 3.2. A state-space representation, therefore, consists of (1) the simultaneous, first-order differential equations from which the state variables can be solved and (2) the algebraic output equation from which all other system variables can be found. A state-space representation is not unique, since a different choice of state variables leads to a different representation of the same system.

In this section, we used two electrical networks to demonstrate some principles that are the foundation of the state-space representation. The representations developed in this section were for single-input, single-output systems, where  $y, D$ , and  $u$  in Eqs. (3.15) and (3.16) are scalar quantities. In general, systems have multiple inputs and multiple outputs. For these cases,  $y$  and  $u$  become vector quantities, and  $D$  becomes a matrix. In Section 3.3 we will generalize the representation for multiple-input, multiple-output systems and summarize the concept of the state-space representation.

### 3.3 The General State-Space Representation

Now that we have represented a physical network in state space and have a good idea of the terminology and the concept, let us summarize and generalize the representation for linear differential equations. First, we formalize some of the definitions that we came across in the last section.

*Linear combination.* A linear combination of  $n$  variables,  $x_i$ , for  $i = 1$  to  $n$ , is given by the following sum,  $S$ :

$$S = K_n x_n + K_{n-1} x_{n-1} + \cdots + K_1 x_1 \quad (3.17)$$

where each  $K_i$  is a constant.

*Linear independence.* A set of variables is said to be linearly independent if none of the variables can be written as a linear combination of the others. For example, given  $x_1$ ,  $x_2$ , and  $x_3$ , if  $x_2 = 5x_1 + 6x_3$ , then the variables are not linearly independent, since one of them can be written as a linear combination of the other two. Now, what must be true so that one variable cannot be written as a linear combination of the other variables? Consider the example  $K_2x_2 = K_1x_1 + K_3x_3$ . If no  $x_i = 0$ , then any  $x_i$  can be written as a linear combination of other variables, unless all  $K_i = 0$ . Formally, then, variables  $x_i$ , for  $i = 1$  to  $n$ , are said to be linearly independent if their linear combination,  $S$ , equals zero *only* if every  $K_i = 0$  and *no*  $x_i = 0$  for all  $t \geq 0$ .

*System variable.* Any variable that responds to an input or initial conditions in a system.

*State variables.* The smallest set of linearly independent system variables such that the values of the members of the set at time  $t_0$  along with known forcing functions completely determine the value of all system variables for all  $t \geq t_0$ .

*State vector.* A vector whose elements are the state variables.

*State space.* The  $n$ -dimensional space whose axes are the state variables. This is a new term and is illustrated in Figure 3.3, where the state variables are assumed to be a resistor voltage,  $v_R$ , and a capacitor voltage,  $v_C$ . These variables form the axes of the *state space*. A trajectory can be thought of as being mapped out by the state vector,  $\mathbf{x}(t)$ , for a range of  $t$ . Also shown is the state vector at the particular time  $t = 4$ .

*State equations.* A set of  $n$  simultaneous, first-order differential equations with  $n$  variables, where the  $n$  variables to be solved are the state variables.

*Output equation.* The algebraic equation that expresses the output variables of a system as linear combinations of the state variables and the inputs.

Now that the definitions have been formally stated, we define the state-space representation of a system. A system is represented in state space by the following equations:

$$\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu} \quad (3.18)$$

$$\mathbf{y} = \mathbf{Cx} + \mathbf{Du} \quad (3.19)$$

for  $t \geq t_0$  and initial conditions,  $\mathbf{x}(t_0)$ , where

$\mathbf{x}$  = state vector

$\dot{\mathbf{x}}$  = derivative of the state vector with respect to time

$\mathbf{y}$  = output vector

$\mathbf{u}$  = input or control vector

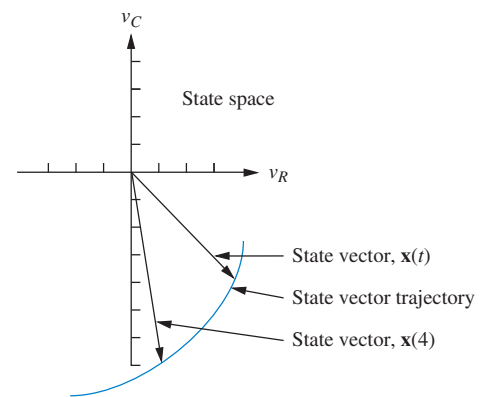
$\mathbf{A}$  = system matrix

$\mathbf{B}$  = input matrix

$\mathbf{C}$  = output matrix

$\mathbf{D}$  = feedforward matrix

Equation (3.18) is called the *state equation*, and the vector  $\mathbf{x}$ , the *state vector*, contains the state variables. Equation (3.18) can be solved for the state variables, which we demonstrate in Chapter 4. Equation (3.19) is called the *output equation*. This equation is used to calculate any other system variables. This representation of a system provides complete knowledge of all variables of the system at any  $t \geq t_0$ .



**FIGURE 3.3** Graphic representation of state space and a state vector

As an example, for a linear, time-invariant, second-order system with a single input  $v(t)$ , the state equations could take on the following form:

$$\frac{dx_1}{dt} = a_{11}x_1 + a_{12}x_2 + b_1v(t) \quad (3.20a)$$

$$\frac{dx_2}{dt} = a_{21}x_1 + a_{22}x_2 + b_2v(t) \quad (3.20b)$$

where  $x_1$  and  $x_2$  are the state variables. If there is a single output, the output equation could take on the following form:

$$y = c_1x_1 + c_2x_2 + d_1v(t) \quad (3.21)$$

The choice of state variables for a given system is not unique. The requirement in choosing the state variables is that they be linearly independent and that a minimum number of them be chosen.

### 3.4 Applying the State-Space Representation

In this section, we apply the state-space formulation to the representation of more complicated physical systems. The first step in representing a system is to select the state vector, which must be chosen according to the following considerations:

1. A minimum number of state variables must be selected as components of the state vector. This minimum number of state variables is sufficient to describe completely the state of the system.
2. The components of the state vector (that is, this minimum number of state variables) must be linearly independent.

Let us review and clarify these statements.

#### Linearly Independent State Variables

The components of the state vector must be linearly independent. For example, following the definition of linear independence in Section 3.3, if  $x_1$ ,  $x_2$ , and  $x_3$  are chosen as state variables, but  $x_3 = 5x_1 + 4x_2$ , then  $x_3$  is not linearly independent of  $x_1$  and  $x_2$ , since knowledge of the values of  $x_1$  and  $x_2$  will yield the value of  $x_3$ . Variables and their successive derivatives are linearly independent. For example, the voltage across an inductor,  $v_L$ , is linearly independent of the current through the inductor,  $i_L$ , since  $v_L = Ldi_L/dt$ . Thus,  $v_L$  cannot be evaluated as a linear combination of the current,  $i_L$ .

#### Minimum Number of State Variables

How do we know the minimum number of state variables to select? Typically, the minimum number required equals the order of the differential equation describing the system. For example, if a third-order differential equation describes the system, then three simultaneous, first-order differential equations are required along with three state variables. From the perspective of the transfer function, the order of the differential equation is the order of the denominator of the transfer function after canceling common factors in the numerator and denominator.

In most cases, another way to determine the number of state variables is to count the number of independent energy-storage elements in the system.<sup>5</sup> The number of

<sup>5</sup> Sometimes it is not apparent in a schematic how many independent energy-storage elements there are. It is possible that more than the minimum number of energy-storage elements could be selected, leading to a state vector whose components number more than the minimum required and are not linearly independent. Selecting additional dependent energy-storage elements results in a system matrix of higher order and more complexity than required for the solution of the state equations.

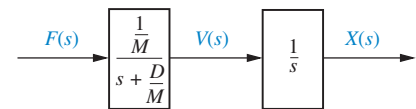


these energy-storage elements equals the order of the differential equation and the number of state variables. In Figure 3.2 there are two energy-storage elements, the capacitor and the inductor. Hence, two state variables and two state equations are required for the system.

If too few state variables are selected, it may be impossible to write particular output equations, since some system variables cannot be written as a linear combination of the reduced number of state variables. In many cases, it may be impossible even to complete the writing of the state equations, since the derivatives of the state variables cannot be expressed as linear combinations of the reduced number of state variables.

If you select the minimum number of state variables but they are not linearly independent, at best you may not be able to solve for all other system variables. At worst you may not be able to complete the writing of the state equations.

Often the state vector includes more than the minimum number of state variables required. Two possible cases exist. Often state variables are chosen to be physical variables of a system, such as position and velocity in a mechanical system. Cases arise where these variables, although linearly independent, are also *decoupled*. That is, some linearly independent variables are not required in order to solve for any of the other linearly independent variables or any other dependent system variable. Consider the case of a mass and viscous damper whose differential equation is  $M dv/dt + Dv = f(t)$ , where  $v$  is the velocity of the mass. Since this is a first-order equation, one state equation is all that is required to define this system in state space with velocity as the state variable. Also, since there is only one energy-storage element, mass, only one state variable is required to represent this system in state space. However, the mass also has an associated position, which is linearly independent of velocity. If we want to include position in the state vector along with velocity, then we add position as a state variable that is linearly independent of the other state variable, velocity. Figure 3.4 illustrates what is happening. The first block is the transfer function equivalent to  $Mdv(t)/dt + Dv(t) = f(t)$ . The second block shows that we integrate the output velocity to yield output displacement (see Table 2.2, Item 10). Thus, if we want displacement as an output, the denominator, or characteristic equation, has increased in order to 2, the product of the two transfer functions. Many times, the writing of the state equations is simplified by including additional state variables.



**FIGURE 3.4** Block diagram of a mass and damper

Another case that increases the size of the state vector arises when the added variable is not linearly independent of the other members of the state vector. This usually occurs when a variable is selected as a state variable but its dependence on the other state variables is not immediately apparent. For example, energy-storage elements may be used to select the state variables, and the dependence of the variable associated with one energy-storage element on the variables of other energy-storage elements may not be recognized. Thus, the dimension of the system matrix is increased unnecessarily, and the solution for the state vector, which we cover in Chapter 4, is more difficult. Also, adding dependent state variables affects the designer's ability to use state-space methods for design.<sup>6</sup>

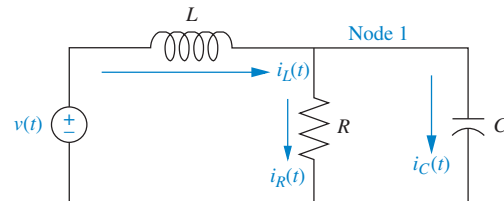
We saw in Section 3.2 that the state-space representation is not unique. The following example demonstrates one technique for selecting state variables and representing a system in state space. Our approach is to write the simple derivative equation for each energy-storage element and solve for each derivative term as a linear combination of any of the system variables and the input that are present in the equation. Next we select each differentiated variable as a state variable. Then we express all other system variables in the equations in terms of the state variables and the input. Finally, we write the output variables as linear combinations of the state variables and the input.

<sup>6</sup>See Chapter 12 for state-space design techniques.

### Example 3.1

#### Representing an Electrical Network

**PROBLEM:** Given the electrical network of Figure 3.5, find a state-space representation if the output is the current through the resistor.



**FIGURE 3.5** Electrical network for representation in state space

**SOLUTION:** The following steps will yield a viable representation of the network in state space.

**Step 1** Label all of the branch currents in the network. These include  $i_L$ ,  $i_R$ , and  $i_C$ , as shown in Figure 3.5.

**Step 2** Select the state variables by writing the derivative equation for all energy-storage elements, that is, the inductor and the capacitor. Thus,

$$C \frac{dv_C}{dt} = i_C \quad (3.22)$$

$$L \frac{di_L}{dt} = v_L \quad (3.23)$$

From Eqs. (3.22) and (3.23), choose the state variables as the quantities that are differentiated, namely  $v_C$  and  $i_L$ . Using Eq. (3.20) as a guide, we see that the state-space representation is complete if the right-hand sides of Eqs. (3.22) and (3.23) can be written as linear combinations of the state variables and the input.

Since  $i_C$  and  $v_L$  are not state variables, our next step is to express  $i_C$  and  $v_L$  as linear combinations of the state variables,  $v_C$  and  $i_L$ , and the input,  $v(t)$ .

**Step 3** Apply network theory, such as Kirchhoff's voltage and current laws, to obtain  $i_C$  and  $v_L$  in terms of the state variables,  $v_C$  and  $i_L$ . At Node 1,

$$\begin{aligned} i_C &= -i_R + i_L \\ &= -\frac{1}{R}v_C + i_L \end{aligned} \quad (3.24)$$

which yields  $i_C$  in terms of the state variables,  $v_C$  and  $i_L$ .

Around the outer loop,

$$v_L = -v_C + v(t) \quad (3.25)$$

which yields  $v_L$  in terms of the state variable,  $v_C$ , and the source,  $v(t)$ .

**Step 4** Substitute the results of Eqs. (3.24) and (3.25) into Eqs. (3.22) and (3.23) to obtain the following state equations:

$$C \frac{dv_C}{dt} = -\frac{1}{R}v_C + i_L \quad (3.26a)$$

$$L \frac{di_L}{dt} = -v_C + v(t) \quad (3.26b)$$

or

$$\frac{dv_C}{dt} = -\frac{1}{RC}v_C + \frac{1}{C}i_L \quad (3.27a)$$

$$\frac{di_L}{dt} = -\frac{1}{L}v_C + \frac{1}{L}v(t) \quad (3.27b)$$

**Step 5** Find the output equation. Since the output is  $i_R(t)$ ,

$$i_R = \frac{1}{R}v_C \quad (3.28)$$

The final result for the state-space representation is found by representing Eqs. (3.27) and (3.28) in vector-matrix form as follows:

$$\begin{bmatrix} \dot{v}_C \\ \dot{i}_L \end{bmatrix} = \begin{bmatrix} -1/(RC) & 1/C \\ -1/L & 0 \end{bmatrix} \begin{bmatrix} v_C \\ i_L \end{bmatrix} + \begin{bmatrix} 0 \\ 1/L \end{bmatrix} v(t) \quad (3.29a)$$

$$i_R = \begin{bmatrix} 1/R & 0 \end{bmatrix} \begin{bmatrix} v_C \\ i_L \end{bmatrix} \quad (3.29b)$$

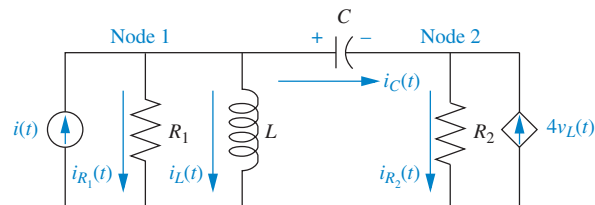
where the dot indicates differentiation with respect to time.

In order to clarify the representation of physical systems in state space, we will look at two more examples. The first is an electrical network with a dependent source. Although we will follow the same procedure as in the previous problem, this problem will yield increased complexity in applying network analysis to find the state equations. For the second example, we find the state-space representation of a mechanical system.

### Example 3.2

#### Representing an Electrical Network with a Dependent Source

**PROBLEM:** Find the state and output equations for the electrical network shown in Figure 3.6 if the output vector is  $\mathbf{y} = [v_{R_2} \quad i_{R_2}]^T$ , where  $T$  means transpose.<sup>7</sup>



**FIGURE 3.6** Electrical network for Example 3.2

**SOLUTION:** Immediately notice that this network has a voltage-dependent current source.

<sup>7</sup>See Appendix G for a discussion of the transpose. Appendix G is located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

**Step 1** Label all of the branch currents on the network, as shown in Figure 3.6.

**Step 2** Select the state variables by listing the voltage-current relationships for all of the energy-storage elements:

$$L \frac{di_L}{dt} = v_L \quad (3.30a)$$

$$C \frac{dv_C}{dt} = i_C \quad (3.30b)$$

From Eqs. (3.30) select the state variables to be the differentiated variables. Thus, the state variables,  $x_1$  and  $x_2$ , are

$$x_1 = i_L; \quad x_2 = v_C \quad (3.31)$$

**Step 3** Remembering that the form of the state equation is

$$\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu} \quad (3.32)$$

we see that the remaining task is to transform the right-hand side of Eq. (3.30) into linear combinations of the state variables and input source current. Using Kirchhoff's voltage and current laws, we find  $v_L$  and  $i_C$  in terms of the state variables and the input current source.

Around the mesh containing  $L$  and  $C$ ,

$$v_L = v_C + v_{R_2} = v_C + i_{R_2} R_2 \quad (3.33)$$

But at Node 2,  $i_{R_2} = i_C + 4v_L$ . Substituting this relationship for  $i_{R_2}$  into Eq. (3.33) yields

$$v_L = v_C + (i_C + 4v_L)R_2 \quad (3.34)$$

Solving for  $v_L$ , we get

$$v_L = \frac{1}{1 - 4R_2}(v_C + i_C R_2) \quad (3.35)$$

Notice that since  $v_C$  is a state variable, we only need to find  $i_C$  in terms of the state variables. We will then have obtained  $v_L$  in terms of the state variables.

Thus, at Node 1 we can write the sum of the currents as

$$\begin{aligned} i_C &= i(t) - i_{R_1} - i_L \\ &= i(t) - \frac{v_{R_1}}{R_1} - i_L \\ &= i(t) - \frac{v_L}{R_1} - i_L \end{aligned} \quad (3.36)$$

where  $v_{R_1} = v_L$ . Equations (3.35) and (3.36) are two equations relating  $v_L$  and  $i_C$  in terms of the state variables  $i_L$  and  $v_C$ . Rewriting Eqs. (3.35) and (3.36), we obtain two simultaneous equations yielding  $v_L$  and  $i_C$  as linear combinations of the state variables  $i_L$  and  $v_C$ :

$$(1 - 4R_2)v_L - R_2 i_C = v_C \quad (3.37a)$$

$$-\frac{1}{R_1}v_L - i_C = i_L - i(t) \quad (3.37b)$$

Solving Eq. (3.37a) simultaneously for  $v_L$  and  $i_C$  yields

$$v_L = \frac{1}{\Delta} [R_2 i_L - v_C - R_2 i(t)] \quad (3.38)$$

and

$$i_C = \frac{1}{\Delta} \left[ (1 - 4R_2) i_L + \frac{1}{R_1} v_C - (1 - 4R_2) i(t) \right] \quad (3.39)$$

where

$$\Delta = - \left[ (1 - 4R_2) + \frac{R_2}{R_1} \right] \quad (3.40)$$

Substituting Eqs. (3.38) and (3.39) into (3.30), simplifying, and writing the result in vector-matrix form renders the following state equation:

$$\begin{aligned} \begin{bmatrix} \dot{i}_L \\ \dot{v}_C \end{bmatrix} &= \begin{bmatrix} R_2/(L\Delta) & -1/(L\Delta) \\ (1 - 4R_2)/(C\Delta) & 1/(R_1 C\Delta) \end{bmatrix} \begin{bmatrix} i_L \\ v_C \end{bmatrix} \\ &+ \begin{bmatrix} -R_2/(L\Delta) \\ -(1 - 4R_2)/(C\Delta) \end{bmatrix} i(t) \end{aligned} \quad (3.41)$$

**Step 4** Derive the output equation. Since the specified output variables are  $v_{R_2}$  and  $i_{R_2}$ , we note that around the mesh containing  $C$ ,  $L$ , and  $R_2$ ,

$$v_{R_2} = -v_C + v_L \quad (3.42a)$$

$$i_{R_2} = i_C + 4v_L \quad (3.42b)$$

Substituting Eqs. (3.38) and (3.39) into Eq. (3.42),  $v_{R_2}$  and  $i_{R_2}$  are obtained as linear combinations of the state variables,  $i_L$  and  $v_C$ . In vector-matrix form, the output equation is

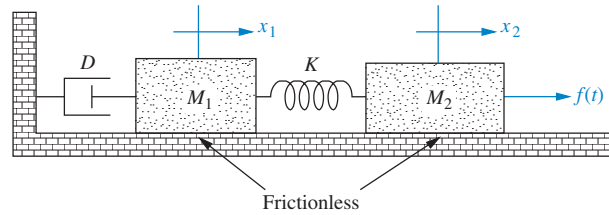
$$\begin{bmatrix} v_{R_2} \\ i_{R_2} \end{bmatrix} = \begin{bmatrix} R_2/\Delta & -(1 + 1/\Delta) \\ 1/\Delta & (1 - 4R_1)/(\Delta R_1) \end{bmatrix} \begin{bmatrix} i_L \\ v_C \end{bmatrix} + \begin{bmatrix} -R_2/\Delta \\ -1/\Delta \end{bmatrix} i(t) \quad (3.43)$$

In the next example, we find the state-space representation for a mechanical system. It is more convenient when working with mechanical systems to obtain the state equations directly from the equations of motion rather than from the energy-storage elements. For example, consider an energy-storage element such as a spring, where  $F = Kx$ . This relationship does not contain the derivative of a physical variable as in the case of electrical networks, where  $i = C dv/dt$  for capacitors, and  $v = L di/dt$  for inductors. Thus, in mechanical systems we change our selection of state variables to be the position and velocity of each point of linearly independent motion. In the example, we will see that although there are three energy-storage elements, there will be four state variables; an additional linearly independent state variable is included for the convenience of writing the state equations. It is left to the student to show that this system yields a fourth-order transfer function if we relate the displacement of either mass to the applied force, and a third-order transfer function if we relate the velocity of either mass to the applied force.

### Example 3.3

#### Representing a Translational Mechanical System

**PROBLEM:** Find the state equations for the translational mechanical system shown in Figure 3.7.



**FIGURE 3.7** Translational mechanical system

**SOLUTION:** First write the differential equations for the network in Figure 3.7, using the methods of Chapter 2 to find the Laplace-transformed equations of motion. Next take the inverse Laplace transform of these equations, assuming zero initial conditions, and obtain

$$M_1 \frac{d^2 x_1}{dt^2} + D \frac{dx_1}{dt} + Kx_1 - Kx_2 = 0 \quad (3.44)$$

$$-Kx_1 + M_2 \frac{d^2 x_2}{dt^2} + Kx_2 = f(t) \quad (3.45)$$

Now let  $d^2 x_1/dt^2 = dv_1/dt$ , and  $d^2 x_2/dt^2 = dv_2/dt$ , and then select  $x_1$ ,  $v_1$ ,  $x_2$ , and  $v_2$  as state variables. Next form two of the state equations by solving Eq. (3.44) for  $dv_1/dt$  and Eq. (3.45) for  $dv_2/dt$ . Finally, add  $dx_1/dt = v_1$  and  $dx_2/dt = v_2$  to complete the set of state equations. Hence,

$$\frac{dx_1}{dt} = \quad \quad \quad +v_1 \quad (3.46a)$$

$$\frac{dv_1}{dt} = -\frac{K}{M_1}x_1 - \frac{D}{M_1}v_1 + \frac{K}{M_1}x_2 \quad (3.46b)$$

$$\frac{dx_2}{dt} = \quad \quad \quad +v_2 \quad (3.46c)$$

$$\frac{dv_2}{dt} = +\frac{K}{M_2}x_1 \quad -\frac{K}{M_2}x_2 \quad +\frac{1}{M_2}f(t) \quad (3.46d)$$

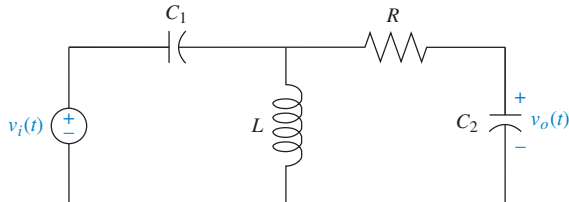
In vector-matrix form,

$$\begin{bmatrix} \dot{x}_1 \\ \dot{v}_1 \\ \dot{x}_2 \\ \dot{v}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -K/M_1 & -D/M_1 & K/M_1 & 0 \\ 0 & 0 & 0 & 1 \\ K/M_2 & 0 & -K/M_2 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ v_1 \\ x_2 \\ v_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1/M_2 \end{bmatrix} f(t) \quad (3.47)$$

where the dot indicates differentiation with respect to time. What is the output equation if the output is  $x(t)$ ?

### Skill-Assessment Exercise 3.1

**PROBLEM:** Find the state-space representation of the electrical network shown in Figure 3.8. The output is  $v_o(t)$ .



**FIGURE 3.8** Electric circuit for Skill-Assessment Exercise 3.1

**ANSWER:**

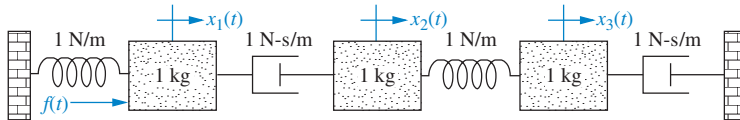
$$\dot{\mathbf{x}} = \begin{bmatrix} 1/C_1 & 1/C_1 & -1/C_1 \\ -1/L & 0 & 0 \\ 1/C_2 & 0 & -1/C_2 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} v_i(t)$$

$$y = [0 \quad 0 \quad 1] \mathbf{x}$$

The complete solution is at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

### Skill-Assessment Exercise 3.2

**PROBLEM:** Represent the translational mechanical system shown in Figure 3.9 in state space, where  $x_3(t)$  is the output.



**FIGURE 3.9** Translational mechanical system for Skill-Assessment Exercise 3.2

**ANSWER:**

$$\dot{\mathbf{z}} = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 \\ -1 & -1 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 1 & -1 & -1 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 0 & 0 & 1 & 0 & -1 & -1 \end{bmatrix} \mathbf{z} + \begin{bmatrix} 0 \\ 1 \\ 0 \\ 0 \\ 0 \\ 0 \end{bmatrix} f(t)$$

$$y = [0 \quad 0 \quad 0 \quad 0 \quad 1 \quad 0] \mathbf{z}$$

where

$$\mathbf{z} = [x_1 \quad \dot{x}_1 \quad x_2 \quad \dot{x}_2 \quad x_3 \quad \dot{x}_3]^T$$

The complete solution is at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

### 3.5 Converting a Transfer Function to State Space

In the last section, we applied the state-space representation to electrical and mechanical systems. We learn how to convert a transfer function representation to a state-space representation in this section. One advantage of the state-space representation is that it can be used for the simulation of physical systems on the digital computer. Thus, if we want to simulate a system that is represented by a transfer function, we must first convert the transfer function representation to state space.

At first we select a set of state variables, called *phase variables*, where each subsequent state variable is defined to be the derivative of the previous state variable. In Chapter 5 we show how to make other choices for the state variables.

Let us begin by showing how to represent a general,  $n$ th-order, linear differential equation with constant coefficients in state space in the phase-variable form. We will then show how to apply this representation to transfer functions.

Consider the differential equation

$$\frac{d^n y}{dt^n} + a_{n-1} \frac{d^{n-1} y}{dt^{n-1}} + \cdots + a_1 \frac{dy}{dt} + a_0 y = b_0 u \quad (3.48)$$

A convenient way to choose state variables is to choose the output,  $y(t)$ , and its  $(n-1)$  derivatives as the state variables. This choice is called the *phase-variable choice*. Choosing the state variables,  $x_i$ , we get

$$x_1 = y \quad (3.49a)$$

$$x_2 = \frac{dy}{dt} \quad (3.49b)$$

$$x_3 = \frac{d^2 y}{dt^2} \quad (3.49c)$$

$$\vdots$$

$$x_n = \frac{d^{n-1} y}{dt^{n-1}} \quad (3.49d)$$

and differentiating both sides yields

$$\dot{x}_1 = \frac{dy}{dt} \quad (3.50a)$$

$$\dot{x}_2 = \frac{d^2 y}{dt^2} \quad (3.50b)$$

$$\dot{x}_3 = \frac{d^3 y}{dt^3} \quad (3.50c)$$

$$\vdots$$

$$\dot{x}_n = \frac{d^n y}{dt^n} \quad (3.50d)$$

where the dot above the  $x$  signifies differentiation with respect to time.

Substituting the definitions of Eq. (3.49) into Eq. (3.50), the state equations are evaluated as



$$\dot{x}_1 = x_2 \quad (3.51a)$$

$$\dot{x}_2 = x_3 \quad (3.51b)$$

$$\vdots$$

$$\dot{x}_{n-1} = x_n \quad (3.51c)$$

$$\dot{x}_n = -a_0x_1 - a_1x_2 \cdots - a_{n-1}x_n + b_0u \quad (3.51d)$$

where Eq. (3.51d) was obtained from Eq. (3.48) by solving for  $d^n y/dt^n$  and using Eq. (3.49). In vector-matrix form, Eq. (3.51) become

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \vdots \\ \dot{x}_{n-1} \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 & 0 & 0 & \cdots & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 & \cdots & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & \cdots & 0 \\ \vdots & & & & & & & \\ 0 & 0 & 0 & 0 & 0 & 0 & \cdots & 1 \\ -a_0 & -a_1 & -a_2 & -a_3 & -a_4 & -a_5 & \cdots & -a_{n-1} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_{n-1} \\ x_n \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 0 \\ \vdots \\ 0 \\ b_0 \end{bmatrix} u \quad (3.52)$$

Equation (3.52) is the phase-variable form of the state equations. This form is easily recognized by the unique pattern of 1's and 0's and the negative of the coefficients of the differential equation written in reverse order in the last row of the system matrix.

Finally, since the solution to the differential equation is  $y(t)$ , or  $x_1$ , the output equation is

$$y = \begin{bmatrix} 1 & 0 & 0 & \cdots & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ \vdots \\ x_{n-1} \\ x_n \end{bmatrix} \quad (3.53)$$

In summary, then, to convert a transfer function into state equations in phase-variable form, we first convert the transfer function to a differential equation by cross-multiplying and taking the inverse Laplace transform, assuming zero initial conditions. Then we represent the differential equation in state space in phase-variable form. An example illustrates the process.

### Example 3.4

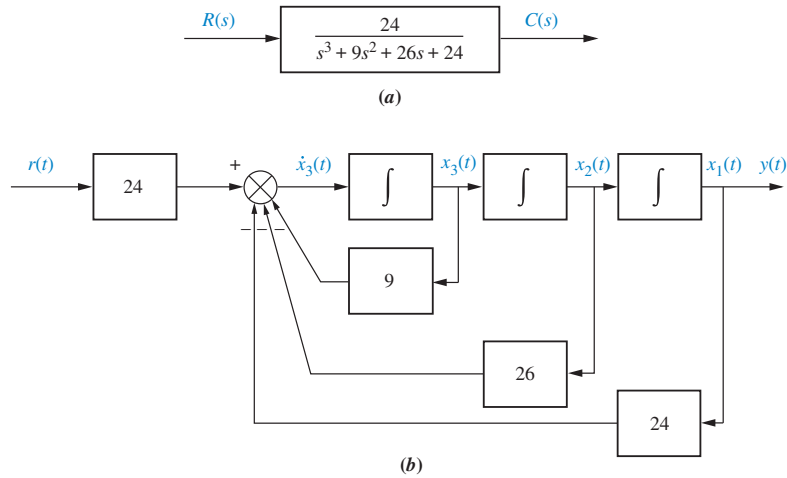
#### Converting a Transfer Function with a Constant Term in the Numerator

**PROBLEM:** Find the state-space representation in phase-variable form for the transfer function shown in Figure 3.10(a).

#### SOLUTION:

**Step 1** Find the associated differential equation. Since

$$\frac{C(s)}{R(s)} = \frac{24}{(s^3 + 9s^2 + 26s + 24)} \quad (3.54)$$



**FIGURE 3.10** a. Transfer function; b. equivalent block diagram showing phase variables. Note:  $y(t) = c(t)$ .

cross-multiplying yields

$$(s^3 + 9s^2 + 26s + 24)C(s) = 24R(s) \tag{3.55}$$

The corresponding differential equation is found by taking the inverse Laplace transform, assuming zero initial conditions:

$$\ddot{c} + 9\dot{c} + 26c + 24c = 24r \tag{3.56}$$

**Step 2** Select the state variables.

Choosing the state variables as successive derivatives, we get

$$x_1 = c \tag{3.57a}$$

$$x_2 = \dot{c} \tag{3.57b}$$

$$x_3 = \ddot{c} \tag{3.57c}$$

Differentiating both sides and making use of Eq. (3.57) to find  $\dot{x}_1$  and  $\dot{x}_2$ , and Eq. (3.56) to find  $\dot{x}_3 = \ddot{c}$ , we obtain the state equations. Since the output is  $c = x_1$ , the combined state and output equations are

$$\dot{x}_1 = x_2 \tag{3.58a}$$

$$\dot{x}_2 = x_3 \tag{3.58b}$$

$$\dot{x}_3 = -24x_1 - 26x_2 - 9x_3 + 24r \tag{3.58c}$$

$$y = c = x_1 \tag{3.58d}$$

In vector-matrix form,

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -24 & -26 & -9 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 24 \end{bmatrix} r \tag{3.59a}$$

$$y = [1 \ 0 \ 0] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} \tag{3.59b}$$

Notice that the third row of the system matrix has the same coefficients as the denominator of the transfer function but negative and in reverse order.

At this point, we can create an equivalent block diagram of the system of Figure 3.10(a) to help visualize the state variables. We draw three integral blocks as shown in Figure 3.10(b) and label each output as one of the state variables,  $x_i(t)$ , as shown. Since the input to each integrator is  $x_i(t)$ , use Eqs. (3.58a), (3.58b), and (3.58c) to determine the combination of input signals to each integrator. Form and label each input. Finally, use Eq. (3.58d) to form and label the output,  $y(t) = c(t)$ . The final result of Figure 3.10(b) is a system equivalent to Figure 3.10(a) that explicitly shows the state variables and gives a vivid picture of the state-space representation.

Students who are using MATLAB should now run ch3p1 through ch3p4 in Appendix B. You will learn how to represent the system matrix  $A$ , the input matrix  $B$ , and the output matrix  $C$  using MATLAB. You will learn how to convert a transfer function to the state-space representation in phase-variable form. Finally, Example 3.4 will be solved using MATLAB.

MATLAB  
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The transfer function of Example 3.4 has a constant term in the numerator. If a transfer function has a polynomial in  $s$  in the numerator that is of order less than the polynomial in the denominator, as shown in Figure 3.11(a), the numerator and denominator can be handled separately. First separate the transfer function into two cascaded transfer functions, as shown in Figure 3.11(b); the first is the denominator, and the second is just the numerator. The first transfer function with just the denominator is converted to the phase-variable representation in state space as demonstrated in the last example. Hence, phase variable  $x_1$  is the output, and the rest of the phase variables are the internal variables of the first block, as shown in Figure 3.11(b). The second transfer function with just the numerator yields

$$Y(s) = C(s) = (b_2s^2 + b_1s + b_0)X_1(s) \quad (3.60)$$

where, after taking the inverse Laplace transform with zero initial conditions,

$$y(t) = b_2 \frac{d^2x_1}{dt^2} + b_1 \frac{dx_1}{dt} + b_0x_1 \quad (3.61)$$

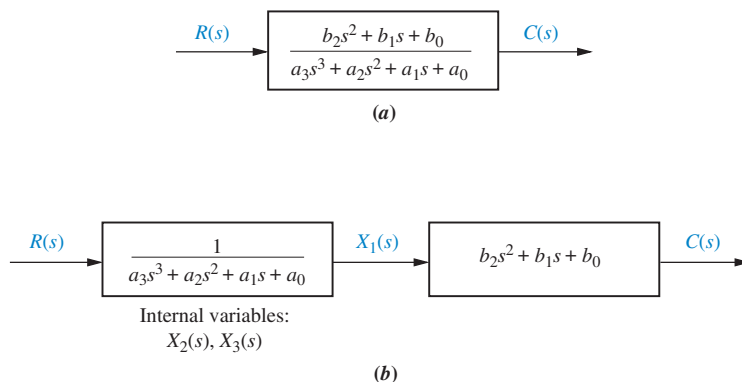


FIGURE 3.11 Decomposing a transfer function

But the derivative terms are the definitions of the phase variables obtained in the first block. Thus, writing the terms in reverse order to conform to an output equation,

$$y(t) = b_0x_1 + b_1x_2 + b_2x_3 \tag{3.62}$$

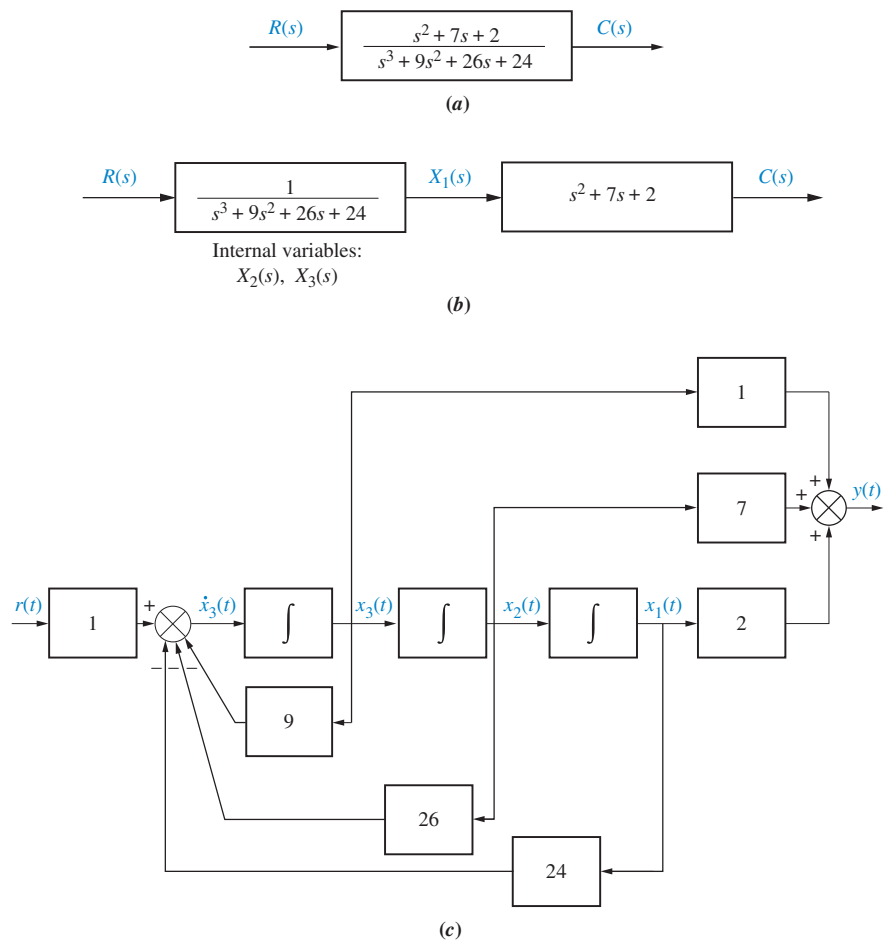
Hence, the second block simply forms a specified linear combination of the state variables developed in the first block.

From another perspective, the denominator of the transfer function yields the state equations, while the numerator yields the output equation. The next example demonstrates the process.

**Example 3.5**

**Converting a Transfer Function with a Polynomial in the Numerator**

**PROBLEM:** Find the state-space representation of the transfer function shown in Figure 3.12(a).



**FIGURE 3.12** a. Transfer function; b. decomposed transfer function; c. equivalent block diagram Note:  $y(t) = c(t)$ .

**SOLUTION:** This problem differs from Example 3.4, since the numerator has a polynomial in  $s$  instead of just a constant term.

**Step 1** Separate the system into two cascaded blocks, as shown in Figure 3.12(b). The first block contains the denominator and the second block contains the numerator.

**Step 2** Find the state equations for the block containing the denominator. We notice that the first block's numerator is  $1/24$  that of Example 3.4. Thus, the state equations are the same except that this system's input matrix is  $1/24$  that of Example 3.4. Hence, the state equation is

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -24 & -26 & -9 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} r \quad (3.63)$$

**Step 3** Introduce the effect of the block with the numerator. The second block of Figure 3.12(b), where  $b_2 = 1$ ,  $b_1 = 7$ , and  $b_0 = 2$ , states that

$$C(s) = (b_2s^2 + b_1s + b_0)X_1(s) = (s^2 + 7s + 2)X_1(s) \quad (3.64)$$

Taking the inverse Laplace transform with zero initial conditions, we get

$$c = \ddot{x}_1 + 7\dot{x}_1 + 2x_1 \quad (3.65)$$

But,

$$\begin{aligned} x_1 &= x_1 \\ \dot{x}_1 &= x_2 \\ \ddot{x}_1 &= x_3 \end{aligned}$$

Hence,

$$y = c(t) = b_2x_3 + b_1x_2 + b_0x_1 = x_3 + x_2 + 2x_1 \quad (3.66)$$

Thus, the last box of Figure 3.11(b) “collects” the states and generates the output equation. From Eq. (3.66),

$$y = [b_0 \quad b_1 \quad b_2] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = [2 \quad 7 \quad 1] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} \quad (3.67)$$

Although the second block of Figure 3.12(b) shows differentiation, this block was implemented without differentiation because of the partitioning that was applied to the transfer function. The last block simply collected derivatives that were already formed by the first block.

Once again we can produce an equivalent block diagram that vividly represents our state-space model. The first block of Figure 3.12(b) is the same as Figure 3.10(a) except for the different constant in the numerator. Thus, in Figure 3.12(c) we reproduce Figure 3.10(b) except for the change in the numerator constant, which appears as a change in the input multiplying factor. The second block of Figure 3.12(b) is represented using Eq. (3.66), which forms the output from a linear combination of the state variables, as shown in Figure 3.12(c).

### TryIt 3.1

Use the following MATLAB statements to form an LTI state-space representation from the transfer function shown in Figure 3.12(a). The **A** matrix and **B** vector are shown in Eq. (3.63). The **C** vector is shown in Eq. (3.67).

```
num=[1 7 2];
den=[1 9 26 24];
[A,B,C,D]=tf2ss...
    (num, den);
P=[0 0 1;0 1 0;1 0 0];
A=inv(P)*A*P
B=inv(P)*B
C=C*P
```

### Skill-Assessment Exercise 3.3

**PROBLEM:** Find the state equations and output equation for the phase-variable representation of the transfer function  $G(s) = \frac{2s + 1}{s^2 + 7s + 9}$ .

**ANSWER:**

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 \\ -9 & -7 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r(t)$$

$$y = [1 \quad 2] \mathbf{x}$$

The complete solution is at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

## 3.6 Converting from State Space to a Transfer Function

In Chapters 2 and 3, we have explored two methods of representing systems: the transfer function representation and the state-space representation. In the last section, we united the two representations by converting transfer functions into state-space representations. Now we move in the opposite direction and convert the state-space representation into a transfer function.

Given the state and output equations

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} \quad (3.68a)$$

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \quad (3.68b)$$

take the Laplace transform assuming zero initial conditions:<sup>8</sup>

$$s\mathbf{X}(s) = \mathbf{A}\mathbf{X}(s) + \mathbf{B}\mathbf{U}(s) \quad (3.69a)$$

$$\mathbf{Y}(s) = \mathbf{C}\mathbf{X}(s) + \mathbf{D}\mathbf{U}(s) \quad (3.69b)$$

Solving for  $\mathbf{X}(s)$  in Eq. (3.69a),

$$(s\mathbf{I} - \mathbf{A})\mathbf{X}(s) = \mathbf{B}\mathbf{U}(s) \quad (3.70)$$

or

$$\mathbf{X}(s) = (s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}\mathbf{U}(s) \quad (3.71)$$

where  $\mathbf{I}$  is the identity matrix.

Substituting Eq. (3.71) into Eq. (3.69b) yields

$$\mathbf{Y}(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}\mathbf{U}(s) + \mathbf{D}\mathbf{U}(s) = [\mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + \mathbf{D}]\mathbf{U}(s) \quad (3.72)$$

We call the matrix  $[\mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B} + \mathbf{D}]$  the transfer function matrix, since it relates the output vector,  $\mathbf{Y}(s)$ , to the input vector,  $\mathbf{U}(s)$ . However, if  $\mathbf{U}(s) = U(s)$  and  $\mathbf{Y}(s) = Y(s)$  are scalars, we can find the transfer function. Thus,

<sup>8</sup>The Laplace transform of a vector is found by taking the Laplace transform of each component. Since  $\dot{\mathbf{x}}$  consists of the derivatives of the state variables, the Laplace transform of  $\dot{\mathbf{x}}$  with zero initial conditions yields each component with the form  $sX_i(s)$ , where  $X_i(s)$  is the Laplace transform of the state variable. Factoring out the complex variable,  $s$ , in each component yields the Laplace transform of  $\dot{\mathbf{x}}$  as  $s\mathbf{X}(s)$ , where  $\mathbf{X}(s)$  is a column vector with components  $X_i(s)$ .

$$T(s) = \frac{Y(s)}{U(s)} = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1} \mathbf{B} + \mathbf{D} \quad (3.73)$$

Let us look at an example.

### Example 3.6

#### State-Space Representation to Transfer Function

**PROBLEM:** Given the system defined by Eq. (3.74), find the transfer function,  $T(s) = Y(s)/U(s)$ , where  $U(s)$  is the input and  $Y(s)$  is the output.

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & -2 & -3 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 10 \\ 0 \\ 0 \end{bmatrix} u \quad (3.74a)$$

$$y = [1 \ 0 \ 0] \mathbf{x} \quad (3.74b)$$

**SOLUTION:** The solution revolves around finding the term  $(s\mathbf{I} - \mathbf{A})^{-1}$  in Eq. (3.73).<sup>9</sup> All other terms are already defined. Hence, first find  $(s\mathbf{I} - \mathbf{A})$ :

$$(s\mathbf{I} - \mathbf{A}) = \begin{bmatrix} s & 0 & 0 \\ 0 & s & 0 \\ 0 & 0 & s \end{bmatrix} - \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & -2 & -3 \end{bmatrix} = \begin{bmatrix} s & -1 & 0 \\ 0 & s & -1 \\ 1 & 2 & s+3 \end{bmatrix} \quad (3.75)$$

Now form  $(s\mathbf{I} - \mathbf{A})^{-1}$ :

$$(s\mathbf{I} - \mathbf{A})^{-1} = \frac{\text{adj}(s\mathbf{I} - \mathbf{A})}{\det(s\mathbf{I} - \mathbf{A})} = \frac{\begin{bmatrix} (s^2 + 3s + 2) & s + 3 & 1 \\ -1 & s(s + 3) & s \\ -s & -(2s + 1) & s^2 \end{bmatrix}}{s^3 + 3s^2 + 2s + 1} \quad (3.76)$$

Substituting  $(s\mathbf{I} - \mathbf{A})^{-1}$ ,  $\mathbf{B}$ ,  $\mathbf{C}$ , and  $\mathbf{D}$  into Eq. (3.73), where

$$\mathbf{B} = \begin{bmatrix} 10 \\ 0 \\ 0 \end{bmatrix}$$

$$\mathbf{C} = [1 \ 0 \ 0]$$

$$\mathbf{D} = 0$$

we obtain the final result for the transfer function:

$$T(s) = \frac{10(s^2 + 3s + 2)}{s^3 + 3s^2 + 2s + 1} \quad (3.77)$$

<sup>9</sup>See Appendix G. It is located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise) and discusses the evaluation of the matrix inverse.

MATLAB  
ML

Students who are using MATLAB should now run `ch3p5` in Appendix B. You will learn how to convert a state-space representation to a transfer function using MATLAB. You can practice by writing a MATLAB program to solve Example 3.6.

Symbolic Math  
SM

Students who are performing the MATLAB exercises and want to explore the added capability of MATLAB's Symbolic Math Toolbox should now run `ch3sp1` in Appendix F located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise). You will learn how to use the Symbolic Math Toolbox to write matrices and vectors. You will see that the Symbolic Math Toolbox yields an alternative way to use MATLAB to solve Example 3.6.

### Skill-Assessment Exercise 3.4

#### TryIt 3.2

Use the following MATLAB and the Control System Toolbox statements to obtain the transfer function shown in Skill-Assessment Exercise 3.4 from the state-space representation of Eq. (3.78).

```
A=[-4 -1.5; 4 0];
B=[2 0]';
C=[1.5 0.625];
D=0;
T=ss(A,B,C,D);
T=tf(T)
```

**PROBLEM:** Convert the state and output equations shown in Eq. (3.78) to a transfer function.

$$\dot{\mathbf{x}} = \begin{bmatrix} -4 & -1.5 \\ 4 & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 2 \\ 0 \end{bmatrix} u(t) \quad (3.78a)$$

$$y = [1.5 \quad 0.625] \mathbf{x} \quad (3.78b)$$

**ANSWER:**

$$G(s) = \frac{3s + 5}{s^2 + 4s + 6}$$

The complete solution is located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

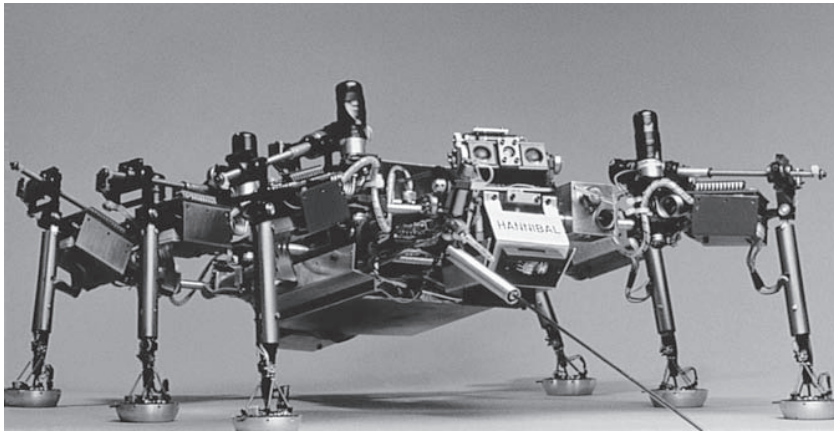
In Example 3.6, the state equations in phase-variable form were converted to transfer functions. In Chapter 5, we will see that other forms besides the phase-variable form can be used to represent a system in state space. The method of finding the transfer function representation for these other forms is the same as that presented in this section.

## 3.7 Linearization

A prime advantage of the state-space representation over the transfer function representation is the ability to represent systems with nonlinearities, such as the one shown in Figure 3.13. The ability to represent nonlinear systems does not imply the ability to solve their state equations for the state variables and the output. Techniques do exist for the solution of some nonlinear state equations, but this study is beyond the scope of this course. However, in Appendix H, located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise), you can see how to use the digital computer to solve state equations. This method also can be used for nonlinear state equations.

If we are interested in small perturbations about an equilibrium point, as we were when we studied linearization in Chapter 2, we can also linearize the state equations about the equilibrium point. The key to linearization about an equilibrium point is, once again, the Taylor series. In the following example, we write the state equations for a simple pendulum, showing that we can represent a nonlinear system in state space; then we linearize the pendulum about its equilibrium point, the vertical position with zero velocity.





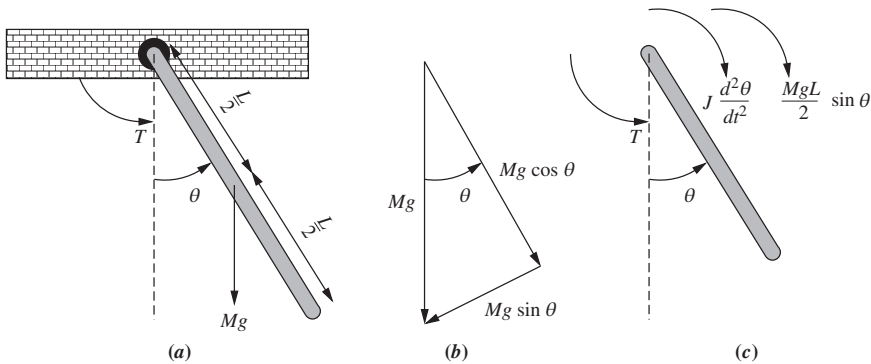
Bruce Frisch/S.S./Photo Researchers, Inc.

**FIGURE 3.13** Walking robots, such as *Hannibal* shown here, can be used to explore hostile environments and rough terrain, such as that found on other planets or inside volcanoes

### Example 3.7

#### Representing a Nonlinear System

**PROBLEM:** First represent the simple pendulum shown in Figure 3.14(a) (which could be a simple model for the leg of the robot shown in Figure 3.13) in state space:  $Mg$  is the weight,  $T$  is an applied torque in the  $\theta$  direction, and  $L$  is the length of the pendulum. Assume the mass is evenly distributed, with the center of mass at  $L/2$ . Then linearize the state equations about the pendulum's equilibrium point—the vertical position with zero angular velocity.



**FIGURE 3.14** a. Simple pendulum; b. force components of  $Mg$ ; c. free-body diagram

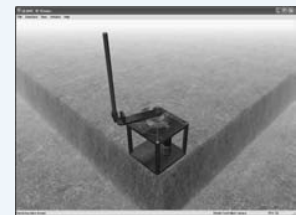
**SOLUTION:** First draw a free-body diagram as shown in Figure 3.14(c). Summing the torques, we get

$$J \frac{d^2 \theta}{dt^2} + \frac{MgL}{2} \sin \theta = T \quad (3.79)$$

where  $J$  is the moment of inertia of the pendulum around the point of rotation. Select the state variables  $x_1$  and  $x_2$  as phase variables. Letting  $x_1 = \theta$  and  $x_2 = d\theta/dt$ , we write the state equations as

#### Virtual Experiment 3.1 Rotary Inverted Pendulum

Put theory into practice by simulating the linear and non-linear model of the Quanser Rotary Inverted Pendulum in LabVIEW. The behavior of an inverted pendulum is similar to a variety of systems, such as Segway transporters and human posture.



Virtual experiments are found on Learning Space.

$$\dot{x}_1 = x_2 \quad (3.80a)$$

$$\dot{x}_2 = -\frac{MgL}{2J} \sin x_1 + \frac{T}{J} \quad (3.80b)$$

where  $\dot{x}_2 = d^2\theta/dt^2$  is evaluated from Eq. (3.79).

Thus, we have represented a nonlinear system in state space. It is interesting to note that the nonlinear Eqs. (3.80) represent a valid and complete model of the pendulum in state space even under nonzero initial conditions and even if parameters are time varying. However, if we want to apply classical techniques and convert these state equations to a transfer function, we must linearize them.

Let us proceed now to linearize the equation about the equilibrium point,  $x_1 = 0, x_2 = 0$ , that is,  $\theta = 0$  and  $d\theta/dt = 0$ . Let  $x_1$  and  $x_2$  be perturbed about the equilibrium point, or

$$x_1 = 0 + \delta x_1 \quad (3.81a)$$

$$x_2 = 0 + \delta x_2 \quad (3.81b)$$

Using Eq. (2.182), we obtain

$$\sin x_1 - \sin 0 = \left. \frac{d(\sin x_1)}{dx_1} \right|_{x_1=0} \delta x_1 = \delta x_1 \quad (3.82)$$

from which

$$\sin x_1 = \delta x_1 \quad (3.83)$$

Substituting Eqs. (3.81) and (3.83) into Eq. (3.80) yields the following state equations:

$$\dot{\delta x}_1 = \delta x_2 \quad (3.84a)$$

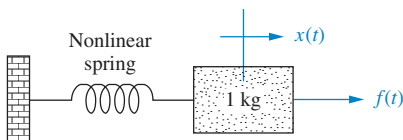
$$\dot{\delta x}_2 = -\frac{MgL}{2J} \delta x_1 + \frac{T}{J} \quad (3.84b)$$

which are linear and a good approximation to Eq. (3.80) for small excursions away from the equilibrium point. What is the output equation?

### Skill-Assessment Exercise 3.5

**PROBLEM:** Represent the translational mechanical system shown in Figure 3.15 in state space about the equilibrium displacement. The spring is nonlinear, where the relationship between the spring force,  $f_s(t)$ , and the spring displacement,  $x_s(t)$ , is  $f_s(t) = 2x_s^2(t)$ . The applied force is  $f(t) = 10 + \delta f(t)$ , where  $\delta f(t)$  is a small force about the 10 N constant value.

Assume the output to be the displacement of the mass,  $x(t)$ .



**FIGURE 3.15** Nonlinear translational mechanical system for Skill-Assessment Exercise 3.5

**ANSWER:**

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 \\ -4\sqrt{5} & 0 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} \delta f(t)$$

$$y = \begin{bmatrix} 1 & 0 \end{bmatrix} \mathbf{x}$$

The complete solution is located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

## Case Studies

### Antenna Control: State-Space Representation

We have covered the state-space representation of individual physical subsystems in this chapter. In Chapter 5, we will assemble individual subsystems into feedback control systems and represent the entire feedback system in state space. Chapter 5 also shows how the state-space representation, via signal-flow diagrams, can be used to interconnect these subsystems and permit the state-space representation of the whole closed-loop system. In the following case study, we look at the antenna azimuth position control system and demonstrate the concepts of this chapter by representing each subsystem in state space.

**PROBLEM:** Find the state-space representation in phase-variable form for each dynamic subsystem in the antenna azimuth position control system shown on the front endpapers, *Configuration 1*. By *dynamic*, we mean that the system does not reach the steady state instantaneously. For example, a system described by a differential equation of first order or higher is a dynamic system. A pure gain, on the other hand, is an example of a nondynamic system, since the steady state is reached instantaneously.

**SOLUTION:** In the case study problem of Chapter 2, each subsystem of the antenna azimuth position control system was identified. We found that the power amplifier and the motor and load were dynamic systems. The preamplifier and the potentiometers are pure gains and so respond instantaneously. Hence, we will find the state-space representations only of the power amplifier and of the motor and load.

#### Power amplifier:

The transfer function of the power amplifier is given on the front endpapers as  $G(s) = 100/(s + 100)$ . We will convert this transfer function to its state-space representation. Letting  $v_p(t)$  represent the power amplifier input and  $e_a(t)$  represent the power amplifier output,

$$G(s) = \frac{E_a(s)}{V_p(s)} = \frac{100}{(s + 100)} \quad (3.85)$$

Cross-multiplying,  $(s + 100)E_a(s) = 100V_p(s)$ , from which the differential equation can be written as

$$\frac{de_a}{dt} + 100e_a = 100v_p(t) \quad (3.86)$$

Rearranging Eq. (3.86) leads to the state equation with  $e_a$  as the state variable:

$$\frac{de_a}{dt} = -100e_a + 100v_p(t) \quad (3.87)$$

Since the output of the power amplifier is  $e_a(t)$ , the output equation is

$$y = e_a \quad (3.88)$$

#### Motor and load:

We now find the state-space representation for the motor and load. We could of course use the motor and load block shown in the block diagram on the front endpapers to obtain the result. However, it is more informative to derive the state-space representation directly from the physics of the motor without first deriving the transfer function. The elements of

the derivation were covered in Section 2.8 but are repeated here for continuity. Starting with Kirchhoff's voltage equation around the armature circuit, we find

$$e_a(t) = i_a(t)R_a + K_b \frac{d\theta_m}{dt} \quad (3.89)$$

where  $e_a(t)$  is the armature input voltage,  $i_a(t)$  is the armature current,  $R_a$  is the armature resistance,  $K_b$  is the armature constant, and  $\theta_m$  is the angular displacement of the armature.

The torque,  $T_m(t)$ , delivered by the motor is related separately to the armature current and the load seen by the armature. From Section 2.8,

$$T_m(t) = K_t i_a(t) = J_m \frac{d^2\theta_m}{dt^2} + D_m \frac{d\theta_m}{dt} \quad (3.90)$$

where  $J_m$  is the equivalent inertia as seen by the armature, and  $D_m$  is the equivalent viscous damping as seen by the armature.

Solving Eq. (3.90) for  $i_a(t)$  and substituting the result into Eq. (3.89) yields

$$e_a(t) = \left( \frac{R_a J_m}{K_t} \right) \frac{d^2\theta_m}{dt^2} + \left( \frac{D_m R_a}{K_t} + K_b \right) \frac{d\theta_m}{dt} \quad (3.91)$$

Defining the state variables  $x_1$  and  $x_2$  as

$$x_1 = \theta_m \quad (3.92a)$$

$$x_2 = \frac{d\theta_m}{dt} \quad (3.92b)$$

and substituting into Eq. (3.91), we get

$$e_a(t) = \left( \frac{R_a J_m}{K_t} \right) \frac{dx_2}{dt} + \left( \frac{D_m R_a}{K_t} + K_b \right) x_2 \quad (3.93)$$

Solving for  $dx_2/dt$  yields

$$\frac{dx_2}{dt} = -\frac{1}{J_m} \left( D_m + \frac{K_t K_b}{R_a} \right) x_2 + \left( \frac{K_t}{R_a J_m} \right) e_a(t) \quad (3.94)$$

Using Eqs. (3.92) and (3.94), the state equations are written as

$$\frac{dx_1}{dt} = x_2 \quad (3.95a)$$

$$\frac{dx_2}{dt} = -\frac{1}{J_m} \left( D_m + \frac{K_t K_b}{R_a} \right) x_2 + \left( \frac{K_t}{R_a J_m} \right) e_a(t) \quad (3.95b)$$

The output,  $\theta_o(t)$ , is 1/10 the displacement of the armature, which is  $x_1$ . Hence, the output equation is

$$y = 0.1x_1 \quad (3.96)$$

In vector-matrix form,

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 \\ 0 & -\frac{1}{J_m} \left( D_m + \frac{K_t K_b}{R_a} \right) \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ \frac{K_t}{R_a J_m} \end{bmatrix} e_a(t) \quad (3.97a)$$

$$y = [0.1 \quad 0] \mathbf{x} \quad (3.97b)$$

But from the case study problem in Chapter 2,  $J_m = 0.03$  and  $D_m = 0.02$ . Also,  $K_i/R_a = 0.0625$  and  $K_b = 0.5$ . Substituting the values into Eq. (3.97a), we obtain the final state-space representation:

$$\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 \\ 0 & -1.71 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 2.083 \end{bmatrix} e_a(t) \quad (3.98a)$$

$$y = [0.1 \quad 0] \mathbf{x} \quad (3.98b)$$

**CHALLENGE:** You are now given a problem to test your knowledge of this chapter's objectives. Referring to the antenna azimuth position control system shown on the front endpapers, find the state-space representation of each dynamic subsystem. Use Configuration 2.

### Pharmaceutical Drug Absorption

An advantage of state-space representation over the transfer function representation is the ability to focus on component parts of a system and write  $n$  simultaneous, first-order differential equations rather than attempt to represent the system as a single,  $n$ th-order differential equation, as we have done with the transfer function. Also, multiple-input, multiple-output systems can be conveniently represented in state space. This case study demonstrates both of these concepts.

**PROBLEM:** In the pharmaceutical industry we want to describe the distribution of a drug in the body. A simple model divides the process into compartments: the dosage, the absorption site, the blood, the peripheral compartment, and the urine. The rate of change of the amount of a drug in a compartment is equal to the input flow rate diminished by the output flow rate. Figure 3.16 summarizes the system. Here each  $x_i$  is the amount of drug in that particular compartment (*Lordi, 1972*). Represent the system in state space, where the outputs are the amounts of drug in each compartment.

**SOLUTION:** The flow rate of the drug into any given compartment is proportional to the concentration of the drug in the previous compartment, and the flow rate out of a given compartment is proportional to the concentration of the drug in its own compartment.

We now write the flow rate for each compartment. The dosage is released to the absorption site at a rate proportional to the dosage concentration, or

$$\frac{dx_1}{dt} = -K_1 x_1 \quad (3.99)$$

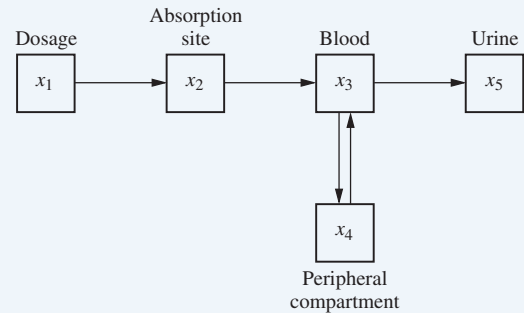
The flow into the absorption site is proportional to the concentration of the drug at the dosage site. The flow from the absorption site into the blood is proportional to the concentration of the drug at the absorption site. Hence,

$$\frac{dx_2}{dt} = K_1 x_1 - K_2 x_2 \quad (3.100)$$

Similarly, the net flow rate into the blood and peripheral compartment is

$$\frac{dx_3}{dt} = K_2 x_2 - K_3 x_3 + K_4 x_4 - K_5 x_3 \quad (3.101)$$

$$\frac{dx_4}{dt} = K_5 x_3 - K_4 x_4 \quad (3.102)$$



**FIGURE 3.16** Pharmaceutical drug-level concentrations in a human

where  $(K_4x_4 - K_5x_3)$  is the net flow rate into the blood from the peripheral compartment. Finally, the amount of the drug in the urine is increased as the blood releases the drug to the urine at a rate proportional to the concentration of the drug in the blood. Thus,

$$\frac{dx_5}{dt} = K_3x_3 \quad (3.103)$$

Equations (3.99) through (3.103) are the state equations. The output equation is a vector that contains each of the amounts,  $x_i$ . Thus, in vector-matrix form,

$$\dot{\mathbf{x}} = \begin{bmatrix} -K_1 & 0 & 0 & 0 & 0 \\ K_1 & -K_2 & 0 & 0 & 0 \\ 0 & K_2 & -(K_3 + K_5) & K_4 & 0 \\ 0 & 0 & K_5 & -K_4 & 0 \\ 0 & 0 & K_3 & 0 & 0 \end{bmatrix} \mathbf{x} \quad (3.104a)$$

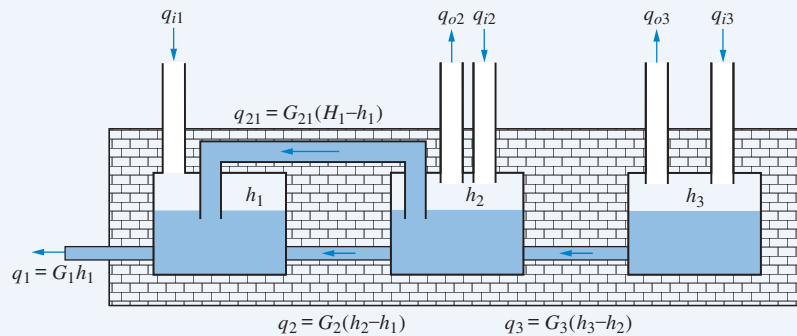
$$\mathbf{y} = \begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 1 \end{bmatrix} \mathbf{x} \quad (3.104b)$$

You may wonder how there can be a solution to these equations if there is no input. In Chapter 4, when we study how to solve the state equations, we will see that initial conditions will yield solutions without forcing functions. For this problem, an initial condition on the amount of dosage,  $x_1$ , will generate drug quantities in all other compartments.

**CHALLENGE:** We now give you a problem to test your knowledge of this chapter's objectives. The problem concerns the storage of water in aquifers. The principles are similar to those used to model pharmaceutical drug absorption.

Underground water supplies, called aquifers, are used in many areas for agricultural, industrial, and residential purposes. An aquifer system consists of a number of interconnected natural storage tanks. Natural water flows through the sand and sandstone of the aquifer system, changing the water levels in the tanks on its way to the sea. A water conservation policy can be established whereby water is pumped between tanks to prevent its loss to the sea.

A model for the aquifer system is shown in Figure 3.17. In this model, the aquifer is represented by three tanks, with water level  $h_i$  called the *head*. Each  $q_n$  is the natural water



**FIGURE 3.17** Aquifer system model

flow to the sea and is proportional to the difference in head between two adjoining tanks, or  $q_n = G_n(h_n - h_{n-1})$ , where  $G_n$  is a constant of proportionality and the units of  $q_n$  are  $\text{m}^3/\text{yr}$ .

The engineered flow consists of three components, also measured in  $\text{m}^3/\text{yr}$ : (1) flow from the tanks for irrigation, industry, and homes,  $q_{on}$ ; (2) replenishing of the tanks from wells,  $q_{in}$ ; and (3) flow,  $q_{21}$ , created by the water conservation policy to prevent loss to the sea. In this model, water for irrigation and industry will be taken only from Tank 2 and Tank 3. Water conservation will take place only between Tank 1 and Tank 2, as follows. Let  $H_1$  be a reference head for Tank 1. If the water level in Tank 1 falls below  $H_1$ , water will be pumped from Tank 2 to Tank 1 to replenish the head. If  $h_1$  is higher than  $H_1$ , water will be pumped back to Tank 2 to prevent loss to the sea. Calling this *flow for conservation*  $q_{21}$ , we can say this flow is proportional to the difference between the head of Tank 1,  $h_1$ , and the reference head,  $H_1$ , or  $q_{21} = G_{21}(H_1 - h_1)$ .

The net flow into a tank is proportional to the rate of change of head in each tank. Thus,

$$C_n dh_n/dt = q_{in} - q_{on} + q_{n+1} - q_n + q_{(n+1)n} - q_{n(n-1)}$$

(Kandel, 1973).

Represent the aquifer system in state space, where the state variables and the outputs are the heads of each tank.

## Summary

This chapter has dealt with the state-space representation of physical systems, which took the form of a state equation,

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} \quad (3.105)$$

and an output equation,

$$\mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{D}\mathbf{u} \quad (3.106)$$

for  $t \geq t_0$ , and initial conditions  $\mathbf{x}(t_0)$ . Vector  $\mathbf{x}$  is called the *state vector* and contains variables, called *state variables*. The state variables can be combined algebraically with the input to form the output equation, Eq. (3.106), from which any other system variables can be found. State variables, which can represent physical quantities such as current or voltage, are chosen to be linearly independent. The choice of state variables is not unique and affects how the matrices  $\mathbf{A}$ ,  $\mathbf{B}$ ,  $\mathbf{C}$ , and  $\mathbf{D}$  look. We will solve the state and output equations for  $\mathbf{x}$  and  $\mathbf{y}$  in Chapter 4.

In this chapter, transfer functions were represented in state space. The form selected was the phase-variable form, which consists of state variables that are successive derivatives of each other. In three-dimensional state space, the resulting system matrix,  $\mathbf{A}$ , for the phase-variable representation is of the form

$$\begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -a_0 & -a_1 & -a_2 \end{bmatrix} \quad (3.107)$$

where the  $a_i$ 's are the coefficients of the characteristic polynomial or denominator of the system transfer function. We also discussed how to convert from a state-space representation to a transfer function.

In conclusion, then, for linear, time-invariant systems, the state-space representation is simply another way of mathematically modeling them. One major advantage of applying

the state-space representation to such linear systems is that it allows computer simulation. Programming the system on the digital computer and watching the system's response is an invaluable analysis and design tool. Simulation is covered in Appendix H located at [www.wiley.com/college/nise](http://www.wiley.com/college/nise).

## Review Questions

1. Give two reasons for modeling systems in state space.
2. State an advantage of the transfer function approach over the state-space approach.
3. Define *state variables*.
4. Define *state*.
5. Define *state vector*.
6. Define *state space*.
7. What is required to represent a system in state space?
8. An eighth-order system would be represented in state space with how many state equations?
9. If the state equations are a system of first-order differential equations whose solution yields the state variables, then the output equation performs what function?
10. What is meant by *linear independence*?
11. What factors influence the choice of state variables in any system?
12. What is a convenient choice of state variables for electrical networks?
13. If an electrical network has three energy-storage elements, is it possible to have a state-space representation with more than three state variables? Explain.
14. What is meant by the phase-variable form of the state equation?

## Problems

1. Represent the electrical network shown in Figure P3.1 in state space, where  $v_o(t)$  is the output. [Section: 3.4]
3. Find the state-space representation of the network shown in Figure P3.3 if the output is  $v_o(t)$ . [Section: 3.4]

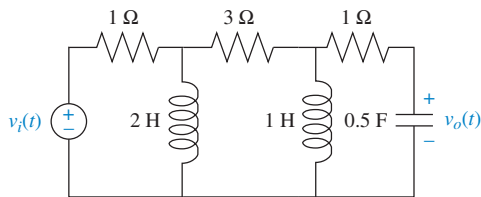


FIGURE P3.1

2. Represent the electrical network shown in Figure P3.2 in state space, where  $i_R(t)$  is the output. [Section: 3.4]

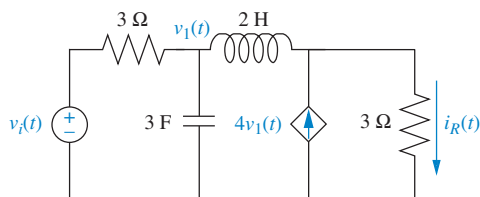


FIGURE P3.2

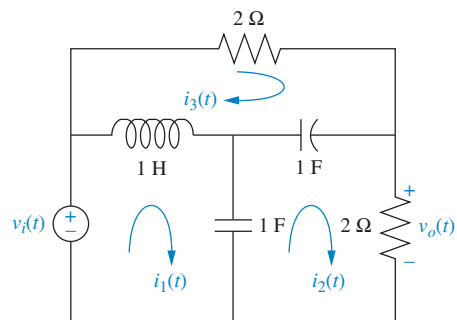


FIGURE P3.3

4. Represent the system shown in Figure P3.4 in state space where the output is  $x_3(t)$ . [Section: 3.4]



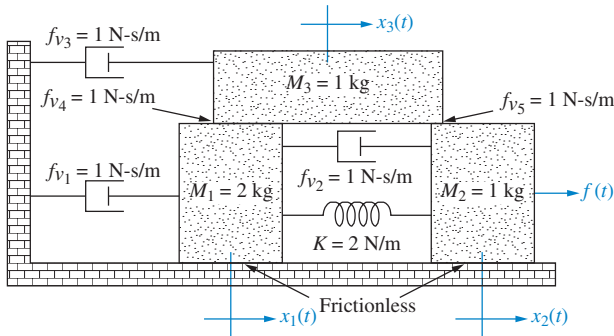


FIGURE P3.4

5. Represent the translational mechanical system shown in Figure P3.5 in state space, where  $x_1(t)$  is the output. [Section: 3.4]

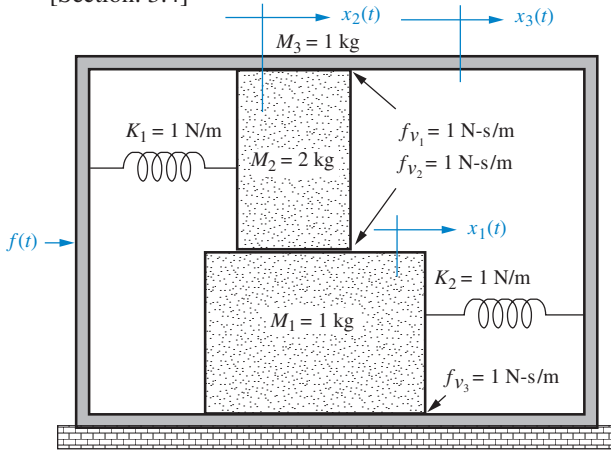


FIGURE P3.5

6. Represent the rotational mechanical system shown in Figure P3.6 in state space, where  $\theta_1(t)$  is the output. [Section: 3.4]

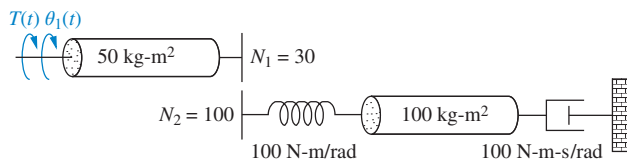


FIGURE P3.6

7. Represent the system shown in Figure P3.7 in state space where the output is  $\theta_L(t)$ . [Section: 3.4]

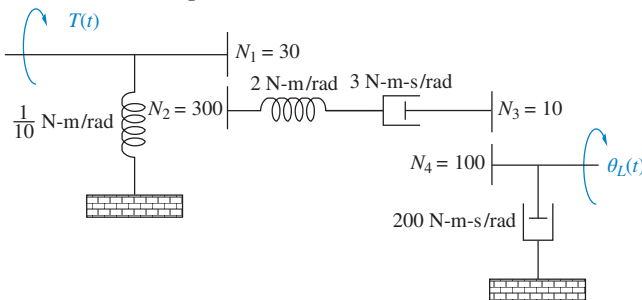


FIGURE P3.7

8. Show that the system of Figure 3.7 in the text yields a fourth-order transfer function if we relate the displacement of either mass to the applied force, and a third-order one if we relate the velocity of either mass to the applied force. [Section: 3.4]
9. Find the state-space representation in phase-variable form for each of the systems shown in Figure P3.8. [Section: 3.5]

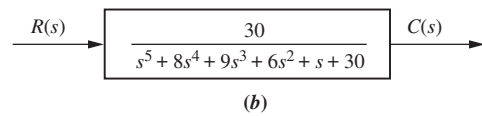
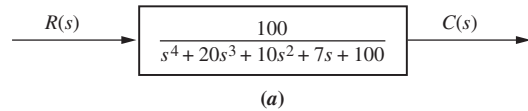


FIGURE P3.8

10. Repeat Problem 9 using MATLAB. [Section: 3.5]

MATLAB  
ML

11. For each system shown in Figure P3.9, write the state equations and the output equation for the phase-variable representation. [Section: 3.5]

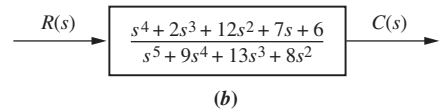
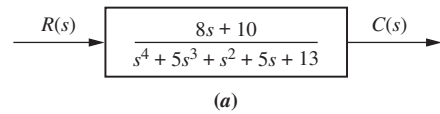


FIGURE P3.9

12. Repeat Problem 11 using MATLAB. [Section: 3.5]

MATLAB  
ML

13. Represent the following transfer function in state space. Give your answer in vector-matrix form. [Section: 3.5]

$$T(s) = \frac{s(s+2)}{(s+1)(s^2+2s+5)}$$

14. Find the transfer function  $G(s) = Y(s)/R(s)$  for each of the following systems represented in state space: [Section: 3.6]

a.  $\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -3 & -2 & -5 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 0 \\ 10 \end{bmatrix} r$

$y = [1 \ 0 \ 0] \mathbf{x}$

b.  $\dot{\mathbf{x}} = \begin{bmatrix} 2 & -3 & -8 \\ 0 & 5 & 3 \\ -3 & -5 & -4 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 1 \\ 4 \\ 6 \end{bmatrix} r$

$y = [1 \ 3 \ 6] \mathbf{x}$

c.  $\dot{\mathbf{x}} = \begin{bmatrix} 3 & -5 & 2 \\ 1 & -8 & 7 \\ -3 & -6 & 2 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 5 \\ -3 \\ 2 \end{bmatrix} r$

$y = [1 \ -4 \ 3] \mathbf{x}$

15. Use MATLAB to find the transfer function,  $G(s)=Y(s)/R(s)$ , for each of the following systems represented in state space: [Section: 3.6]

MATLAB  
ML

a.  $\dot{\mathbf{x}} = \begin{bmatrix} 0 & 1 & 5 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -7 & -9 & -2 & -3 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 0 \\ 5 \\ 8 \\ 2 \end{bmatrix} r$

$y = [1 \ 3 \ 6 \ 6] \mathbf{x}$

b.  $\dot{\mathbf{x}} = \begin{bmatrix} 3 & 1 & 0 & 4 & -2 \\ -3 & 5 & -5 & 2 & -1 \\ 0 & 1 & -1 & 2 & 8 \\ -7 & 6 & -3 & -4 & 0 \\ -6 & 0 & 4 & -3 & 1 \end{bmatrix} \mathbf{x} + \begin{bmatrix} 2 \\ 7 \\ 8 \\ 5 \\ 4 \end{bmatrix} r$

$y = [1 \ -2 \ -9 \ 7 \ 6] \mathbf{x}$

16. Repeat Problem 15 using MATLAB, the Symbolic Math Toolbox, and Eq. (3.73). [Section: 3.6]

Symbolic Math  
SM

17. A missile in flight, as shown in Figure P3.10, is subject to four forces: thrust, lift, drag, and gravity. The missile flies at an angle of attack,  $\alpha$ , from its longitudinal axis, creating lift. For steering, the body angle from vertical,  $\phi$ , is controlled by rotating the engine at the tail. The transfer function relating the body angle,  $\phi$ , to the angular displacement,  $\delta$ , of the engine is of the form

$$\frac{\Phi(s)}{\delta(s)} = \frac{K_a s + K_b}{K_3 s^3 + K_2 s^2 + K_1 s + K_0}$$

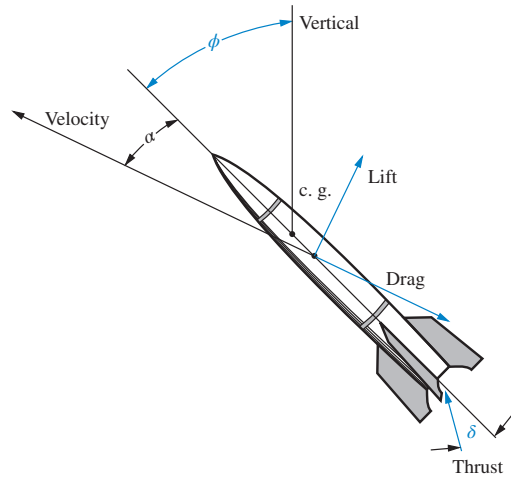


FIGURE P3.10 Missile

Represent the missile steering control in state space. [Section: 3.5]

18. Given the dc servomotor and load shown in Figure P3.11, represent the system in state space, where the state variables are the armature current,  $i_a$ , load displacement,  $\theta_L$ , and load angular velocity,  $\omega_L$ . Assume that the output is the angular displacement of the armature. Do not neglect armature inductance. [Section: 3.4]

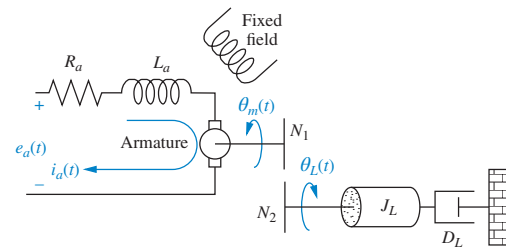


FIGURE P3.11 Motor and load

19. Consider the mechanical system of Figure P3.12. If the spring is nonlinear, and the force,  $F_s$ , required to stretch the spring is  $F_s = 2x_1^2$ , represent the system in state space linearized about  $x_1 = 1$  if the output is  $x_2$ . [Section: 3.7]

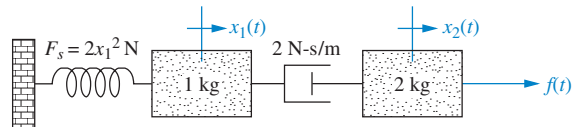
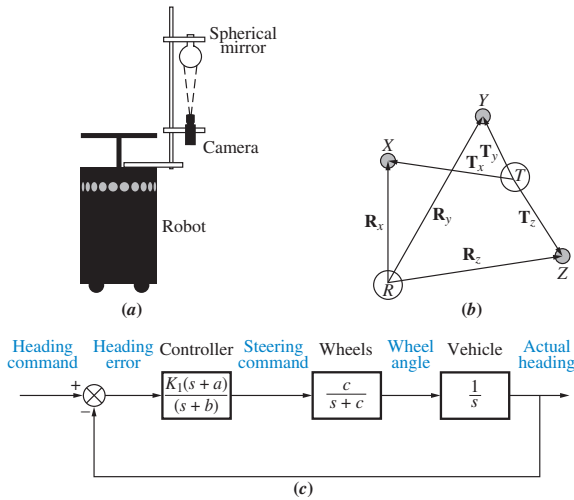


FIGURE P3.12 Nonlinear mechanical system

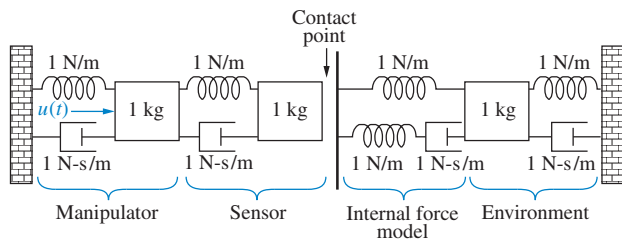
20. Image-based homing for robots can be implemented by generating heading command inputs to a steering system based on the following guidance algorithm. Suppose the robot shown in Figure P3.13(a) is to go from point  $R$  to a target, point  $T$ , as shown in Figure P3.13(b). If  $\mathbf{R}_x$ ,  $\mathbf{R}_y$ , and  $\mathbf{R}_z$  are vectors from the robot to each landmark,  $X$ ,  $Y$ ,  $Z$ , respectively, and  $\mathbf{T}_x$ ,  $\mathbf{T}_y$ , and  $\mathbf{T}_z$  are vectors from the target to each landmark, respectively, then heading commands

would drive the robot to minimize  $\mathbf{R}_x - \mathbf{T}_x$ ,  $\mathbf{R}_y - \mathbf{T}_y$ , and  $\mathbf{R}_z - \mathbf{T}_z$  simultaneously, since the differences will be zero when the robot arrives at the target (Hong, 1992). If Figure P3.13(c) represents the control system that steers the robot, represent each block—the controller, wheels, and vehicle—in state space. An animation PowerPoint presentation (PPT) demonstrating this system is available for instructors at [www.wiley.com/college/nise](http://www.wiley.com/college/nise). See *Robot*. [Section: 3.5]



**FIGURE P3.13** a. Robot with television imaging system;<sup>10</sup> b. vector diagram showing concept behind image-based homing;<sup>10</sup> c. heading control system

21. Modern robotic manipulators that act directly upon their target environments must be controlled so that impact forces as well as steady-state forces do not damage the targets. At the same time, the manipulator must provide sufficient force to perform the task. In order to develop a control system to regulate these forces, the robotic manipulator and target environment must be modeled. Assuming the model shown in Figure P3.14, represent in state space the manipulator and its environment under the following conditions (Chiu, 1997). [Section: 3.5]



**FIGURE P3.14** Robotic manipulator and target environment<sup>11</sup>

<sup>10</sup> Hong, J.; Tan, X.; Pinette, B.; Weiss, R.; and Riseman, E. M. Image-Based Homing, *IEEE Control Systems*, Feb. 1992, pp. 38–45. © 1992 IEEE.

<sup>11</sup> Based on Chiu, D. K., and Lee, S. Design and Experimentation of a Jump Impact Controller. *IEEE Control Systems*, June 1997, Figure 1, p. 99. 1997 IEEE.

- a. The manipulator is not in contact with its target environment.
- b. The manipulator is in constant contact with its target environment.

22. In the past, Type-1 diabetes patients had to inject themselves with insulin three to four times a day. New delayed-action insulin analogues such as insulin Glargine require a single daily dose. A similar procedure to the one described in the Pharmaceutical Drug Absorption case study of this chapter is used to find a model for the concentration-time evolution of plasma for insulin Glargine. For a specific patient, state-space model matrices are given by (Tarín, 2007)

$$\mathbf{A} = \begin{bmatrix} -0.435 & 0.209 & 0.02 \\ 0.268 & -0.394 & 0 \\ 0.227 & 0 & -0.02 \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix};$$

$$\mathbf{C} = [0.0003 \quad 0 \quad 0]; \quad \mathbf{D} = 0$$

where the state vector is given by

$$\mathbf{x} = \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}.$$

The state variables are

- $x_1$  = insulin amount in plasma compartment
- $x_2$  = insulin amount in liver compartment
- $x_3$  = insulin amount in interstitial (in body tissue) compartment

The system's input is  $u$  = external insulin flow. The system's output is  $y$  = plasma insulin concentration.

- a. Find the system's transfer function.
- b. Verify your result using MATLAB.

MATLAB  
ML

23. A linear, time-invariant model of the hypothalamic-pituitary-adrenal axis of the endocrine system with five state variables has been proposed as follows (Kyrylov, 2005):

$$\frac{dx_0}{dt} = a_{00}x_0 + a_{02}x_2 + d_0$$

$$\frac{dx_1}{dt} = a_{10}x_0 + a_{11}x_1 + a_{12}x_2$$

$$\frac{dx_2}{dt} = a_{20}x_0 + a_{21}x_1 + a_{22}x_2 + a_{23}x_3 + a_{24}x_4$$

$$\frac{dx_3}{dt} = a_{32}x_2 + a_{33}x_3$$

$$\frac{dx_4}{dt} = a_{42}x_2 + a_{44}x_4$$

where each of the state variables represents circulatory concentrations as follows:

- $x_0$  = corticotropin-releasing hormone
- $x_1$  = corticotropin
- $x_2$  = free cortisol
- $x_3$  = albumin-bound cortisol
- $x_4$  = corticosteroid-binding globulin
- $d_0$  = an external generating factor

Express the system in the form  $\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu}$ .

24. In this chapter, we described the state-space representation of single-input, single-output systems. In general, systems can have multiple inputs and multiple outputs. An autopilot is to be designed for a submarine as shown in Figure P3.15 to maintain a constant depth under severe wave disturbances. We will see that this system has two inputs and two outputs and thus the scalar  $u$  becomes a vector,  $\mathbf{u}$ , and the scalar  $y$  becomes a vector,  $\mathbf{y}$ , in the state equations.

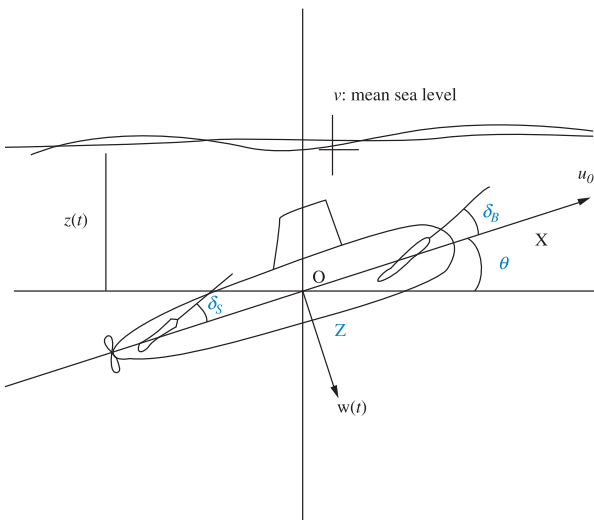


FIGURE P3.15<sup>12</sup>

It has been shown that the system's linearized dynamics under neutral buoyancy and at a given constant speed are given by (Liceaga-Castro, 2009):

$$\begin{aligned} \dot{\mathbf{x}} &= \mathbf{Ax} + \mathbf{Bu} \\ \mathbf{y} &= \mathbf{Cx} \end{aligned}$$

where

$$\mathbf{x} = \begin{bmatrix} w \\ q \\ z \\ \theta \end{bmatrix}; \quad \mathbf{y} = \begin{bmatrix} z \\ \theta \end{bmatrix}; \quad \mathbf{u} = \begin{bmatrix} \delta_B \\ \delta_S \end{bmatrix}$$

<sup>12</sup>Liceaga-Castro E., van der Molen G.M. Submarine  $H^\infty$  Depth Control Under Wave Disturbances. *IEEE Trans. on Control Systems Technology*, Vol. 3 No. 3, 1995. Figure 1, p. 339.

$$\mathbf{A} = \begin{bmatrix} -0.038 & 0.896 & 0 & 0.0015 \\ 0.0017 & -0.092 & 0 & -0.0056 \\ 1 & 0 & 0 & -3.086 \\ 0 & 1 & 0 & 0 \end{bmatrix};$$

$$\mathbf{B} = \begin{bmatrix} -0.0075 & -0.023 \\ 0.0017 & -0.0022 \\ 0 & 0 \\ 0 & 0 \end{bmatrix}; \quad \mathbf{C} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

and where

- $w$  = the heave velocity
- $q$  = the pitch rate
- $z$  = the submarine depth
- $\theta$  = the pitch angle
- $\delta_B$  = the bow hydroplane angle
- $\delta_S$  = the stern hydroplane angle

Since this system has two inputs and two outputs, four transfer functions are possible.

- a. Use MATLAB to calculate the system's matrix transfer function. MATLAB  
ML
- b. Using the results from Part a, write the transfer function  $\frac{z(s)}{\delta_B(s)}$ ,  $\frac{z(s)}{\delta_S(s)}$ ,  $\frac{\theta(s)}{\delta_B(s)}$ , and  $\frac{\theta(s)}{\delta_S(s)}$ .

25. Experiments to identify precision grip dynamics between the index finger and thumb have been performed using a ball-drop experiment. A subject holds a device with a small receptacle into which an object is dropped, and the response is measured (Fagergren, 2000). Assuming a step input, it has been found that the response of the motor subsystem together with the sensory system is of the form

$$G(s) = \frac{Y(s)}{R(s)} = \frac{s + c}{(s^2 + as + b)(s + d)}$$

Convert this transfer function to a state-space representation.

26. State-space representations are, in general, not unique. One system can be represented in several possible ways. For example, consider the following systems:

- a.  $\dot{x} = -5x + 3u$   
 $y = 7x$

b. 
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -5 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 3 \\ 1 \end{bmatrix} u$$

$$y = \begin{bmatrix} 7 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

c. 
$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -5 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 3 \\ 0 \end{bmatrix} u$$

$$y = \begin{bmatrix} 7 & 3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

Show that these systems will result in the same transfer function. We will explore this phenomenon in more detail in Chapter 5.

27. Figure P3.16 shows a schematic description of the global carbon cycle (Li, ). In the figure,  $m_A(t)$  represents the amount of carbon in gigatons (GtC) present in the atmosphere of earth;  $m_V(t)$  the amount in vegetation;  $m_S(t)$  the amount in soil;  $m_{SO}(t)$  the amount in surface ocean; and  $m_{IDO}(t)$  the amount in intermediate and deep-ocean reservoirs. Let  $u_E(t)$  stand for the human generated CO<sub>2</sub> emissions (GtC/yr). From the figure, the atmospheric mass balance in the atmosphere can be expressed as:

$$\frac{dm_A}{dt}(t) = u_E(t) - (k_{O1} + k_{L1})m_A(t) + k_{L2}m_V(t) + k_{O2}m_{SO}(t) + k_{L4}m_S(t)$$

where the k's are exchange coefficients (yr<sup>-1</sup>).

- a. Write the remaining reservoir mass balances. Namely, write equations for  $\frac{dm_{SO}(t)}{dt}$ ,  $\frac{dm_{IDO}(t)}{dt}$ ,  $\frac{dm_V(t)}{dt}$ , and  $\frac{dm_S(t)}{dt}$
- b. Express the system in state-space form.

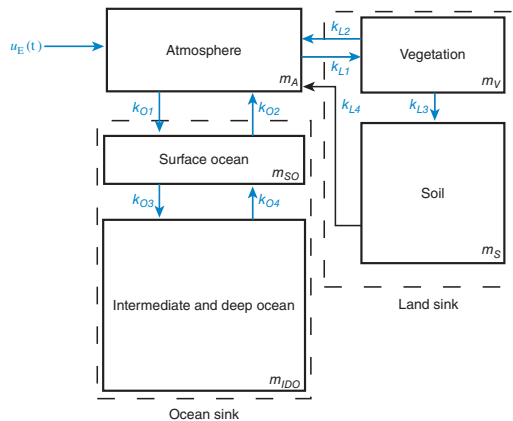


FIGURE P3.16 Global carbon cycle<sup>13</sup>

<sup>13</sup> Li, S., Jarvis, A.J., and Leedal, D.T. Are response function representations of the global carbon cycle ever interpretable? *Tellus*, vol. 61B, 2009, pp. 361–371. (Fig. 1 p. 363).

28. Given the photovoltaic system described in Problem 65 in Chapter 2 (Agee, 2012) and defining the following state variables, system input and output as  $y = x_1 = \theta_m$ ,  $x_2 = \dot{\theta}_m$ ,  $x_3 = i_a$ , and  $u = e_a$ , write a state-space representation of the system in the form  $\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu}$ ,  $\mathbf{y} = \mathbf{Cx}$ .
29. A single-pole oil cylinder valve contains a spool that regulates hydraulic pressure, which is then applied to a piston that drives a load. The transfer function relating piston displacement,  $X_p(s)$  to spool displacement from equilibrium,  $X_v(s)$ , is given by (Qu, 2010):

$$G(s) = \frac{X_p(s)}{X_v(s)} = \frac{K_q \omega_h^2 / A_1}{s(s^2 + 2\zeta \omega_h s + \omega_h^2)}$$

where  $A_1$  = effective area of a the valve's chamber,  $K_q$  = rate of change of the load flow rate with a change in displacement, and  $\omega_h$  = the natural frequency of the hydraulic system. Find the state-space representation of the system, where the state variables are the phase variables associated with the piston.

30. Figure P3.17 shows a free-body diagram of an inverted pendulum, mounted on a cart with a mass,  $M$ . The pendulum has a point mass,  $m$ , concentrated at the upper end of a rod with zero mass, a length,  $l$ , and a frictionless hinge. A motor drives the cart, applying a horizontal force,  $u(t)$ . A gravity force,  $mg$ , acts on  $m$  at all times. The pendulum angle relative to the  $y$ -axis,  $\theta$ , its angular speed,  $\dot{\theta}$ , the horizontal position of the cart,  $x$ , and its speed,  $\dot{x}$ , were selected to be the state variables. The state-space equations derived were heavily nonlinear.<sup>14</sup> They were then linearized around the stationary point,  $\mathbf{x}_0 = \mathbf{0}$  and  $u_0 = 0$ , and manipulated to yield the following open-loop model written in perturbation form:

$$\frac{d}{dt} \delta \mathbf{x} = \mathbf{A} \delta \mathbf{x} + \mathbf{B} \delta u$$

However, since  $\mathbf{x}_0 = \mathbf{0}$  and  $u_0 = 0$ , then let:  $\mathbf{x} = \mathbf{x}_0 + \delta \mathbf{x} = \delta \mathbf{x}$  and  $u = u_0 + \delta u = \delta u$ . Thus the state equation may be rewritten as (Prasad, 2012):

$$\dot{\mathbf{x}} = \mathbf{Ax} + \mathbf{Bu}$$

where

$$\mathbf{A} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ \frac{(M+m)g}{Ml} & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ -\frac{mg}{M} & 0 & 0 & 0 \end{bmatrix} \text{ and } \mathbf{B} = \begin{bmatrix} 0 \\ -1 \\ \frac{Ml}{M} \\ \frac{1}{M} \end{bmatrix}$$

<sup>14</sup> As noted in the introduction to Section 3.7, the techniques for solving such nonlinear state equations are beyond the scope of this course.

Assuming the output to be the horizontal position of  $m = x_m = x + l \sin \theta = x + l\theta$  for a small angle,  $\theta$ , the output equation becomes:

$$y = l\theta + x = \mathbf{C}\mathbf{x} = \begin{bmatrix} l & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \dot{\theta} \\ x \\ \dot{x} \end{bmatrix}$$

Given that:  $M = 2.4 \text{ kg}$ ,  $m = 0.23 \text{ kg}$ ,  $l = 0.36 \text{ m}$ ,  $g = 9.81 \text{ m/s}^2$ , use MATLAB to find the transfer function,  $G(s) = Y(s)/U(s) = X_m(s)/U(s)$ .

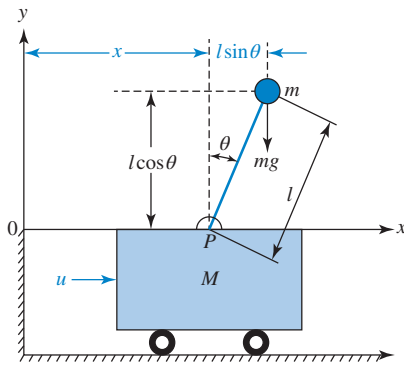


FIGURE P3.17 Motor-driven inverted pendulum cart system<sup>15</sup>

**PROGRESSIVE ANALYSIS AND DESIGN PROBLEMS**

**31. Control of HIV/AIDS.** Problem 67 in Chapter 2 introduced a model for HIV infection. If retroviral drugs, RTIs and PIs as discussed in Problem 22 in Chapter 1, are used, the model is modified as follows (Craig, 2004):

$$\frac{dT}{dt} = s - dT - (1 - u_1) \beta T v$$

$$\frac{dT^*}{dt} = (1 - u_1) \beta T v - \mu T^*$$

$$\frac{dv}{dt} = (1 - u_2) k T^* - cv$$

where  $0 \leq u_1 \leq 1$ ,  $0 \leq u_2 \leq 1$  represent the effectiveness of the RTI and PI medication, respectively.

a. Obtain a state-space representation of the HIV/AIDS model by linearizing the equations about the

$$(T_0, T_0^*, v_0) = \left( \frac{c\mu}{\beta k}, \frac{s}{\mu} - \frac{cd}{\beta k}, \frac{sk}{c\mu} - \frac{d}{\beta} \right)$$

equilibrium with  $u_{10} = u_{20} = 0$ . This equilibrium represents the asymptomatic HIV-infected patient. Note that each one of the above equations is of the form  $\dot{x}_i = f_i(x_i, u_1, u_2)$ ,  $i = 1, 2, 3$ .

b. If Matrices **A** and **B** are given by

$$\mathbf{A} = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \frac{\partial f_1}{\partial x_2} & \frac{\partial f_1}{\partial x_3} \\ \frac{\partial f_2}{\partial x_1} & \frac{\partial f_2}{\partial x_2} & \frac{\partial f_2}{\partial x_3} \\ \frac{\partial f_3}{\partial x_1} & \frac{\partial f_3}{\partial x_2} & \frac{\partial f_3}{\partial x_3} \end{bmatrix}_{T_0, T_0^*, v_0}; \quad \mathbf{B} = \begin{bmatrix} \frac{\partial f_1}{\partial u_1} & \frac{\partial f_1}{\partial u_2} \\ \frac{\partial f_2}{\partial u_1} & \frac{\partial f_2}{\partial u_2} \\ \frac{\partial f_3}{\partial u_1} & \frac{\partial f_3}{\partial u_2} \end{bmatrix}_{T_0, T_0^*, v_0}$$

and we are interested in the number of free HIV viruses as the system's output,

$$\mathbf{C} = \begin{bmatrix} 0 & 0 & 1 \end{bmatrix}$$

show that

$$\mathbf{A} = \begin{bmatrix} -(d + \beta v_0) & 0 & -\beta T_0 \\ \beta v_0 & -\mu & \beta T_0 \\ 0 & k & -c \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} \beta T_0 v_0 & 0 \\ -\beta T_0 v_0 & 0 \\ 0 & -k T_0^* \end{bmatrix}$$

c. Typical parameter values and descriptions for the HIV/AIDS model are shown in the following table. Substitute the values from the table into your model and write as

$$\begin{aligned} \dot{\mathbf{x}} &= \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} \\ \mathbf{y} &= \mathbf{C}\mathbf{x} \end{aligned}$$

Table of HIV/AIDS Model Parameters<sup>16</sup>

$t$	Time	days
$d$	Death of uninfected T cells	0.02/day
$k$	Rate of free viruses produced per infected T cell	100 counts/cell
$s$	Source term for uninfected T cells	10/mm <sup>3</sup> /day
$\beta$	Infectivity rate of free virus particles	2.4 × 10 <sup>-5</sup> /mm <sup>3</sup> /day
$c$	Death rate of viruses	2.4/day
$\mu$	Death rate of infected T cells	0.24/day

**32. Hybrid vehicle.** For Problem 23 in Chapter 1 we developed the functional block diagrams for the cruise control of serial, parallel, and split-power

<sup>15</sup> Prasad, L., Tyagi, B., and Gupta, H. Modeling & Simulation for Optimal Control of Nonlinear Inverted Pendulum Dynamical System using PID Controller & LQR. *IEEE Computer Society Sixth Asia Modeling Symposium*, 2012, pp. 138–143. Figure 1 p. 139. Reproduced with permission of IEEE in the format Republish in a book via Copyright Clearance Center.

<sup>16</sup> Craig, I. K., Xia, X., and Venter, J. W. Introducing HIV/AIDS Education Into the Electrical Engineering Curriculum at the University of Pretoria. *IEEE Transactions on Education*, vol. 47, no. 1, February 2004, pp. 65–73. Table II, p. 67. Modelling Symposium (AMS), 2012 Sixth Asia by IEEE. Reproduced with permission of IEEE in the format Republish in a book via Copyright Clearance Center.

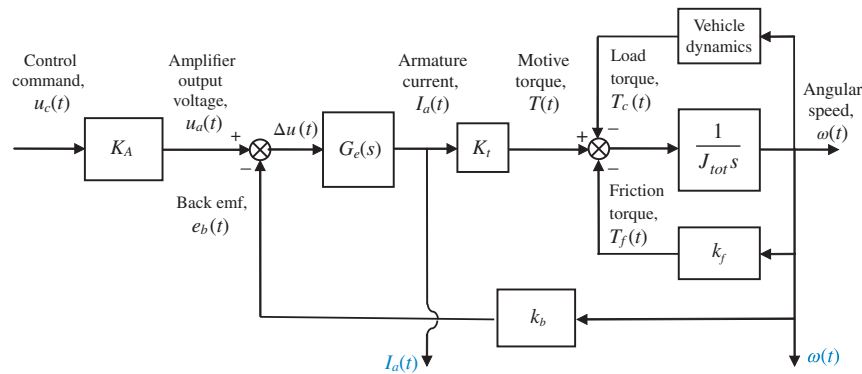


FIGURE P3.18 Block diagram representation of an HEV forward path<sup>17</sup>

hybrid electric vehicles (HEV). Those diagrams showed that the engine or electric motor or both may propel the vehicle. When electric motors are the sole providers of the motive force, the forward paths of all HEV topologies are similar. In general, such a forward path can be represented (Preitl, 2007) by a block diagram similar to the one of Figure P3.18.

Assume the motor to be an armature-controlled dc motor. In this diagram,  $K_A$  is the power amplifier gain;  $G_e(s)$  is the transfer function of the motor electric circuit and consists of a series inductor and resistor,  $L_a$  and  $R_a$ , respectively;  $K_t$  is the motor torque constant;  $J_{tot}$  is the sum of the motor inertia,  $J_m$ , the inertias of the vehicle,  $J_{veh}$ , and the two driven wheels,  $J_w$ , both of which are reflected to the motor shaft;  $k_f$  is the coefficient of viscous friction; and  $k_b$  is the back emf constant.

The input variables are  $u_c(t)$ , the command voltage from the electronic control unit and  $T_c(t)$ , the load torque. The output variables in this block diagram are the motor angular speed,  $\omega(t)$ , and its armature current,  $I_a(t)$ .

- a. Write the basic time-domain equations that characterize the relationships between the state, input, and output variables for the block diagram of Figure P3.18, given that the state variables are the motor armature current,  $I_a(t)$ , and angular speed,  $\omega(t)$ .
- b. Write the resulting state-space equations and then represent them in matrix form. Regard the load torque  $T_c(t)$  as an extra input to the system. Thus, in your resulting state-space representation, the system will have two inputs and two outputs.

33. **Parabolic trough collector.** A transfer function model from fluid flow to fluid temperature for a parabolic trough collector was introduced in Problem 69, Chapter 2. A more detailed model for the response of this system is given under specific operation conditions (Camacho, 2012) by:

$$\frac{H}{Q}(s) = \frac{137.2 \times 10^{-6}}{s^2 + 0.0224s + 196 \times 10^{-6}} e^{-39s}$$

Find an appropriate state-space representation for the system.

<sup>17</sup> Preitl, Z., Bauer, P., and J. Bokor, J. A Simple Control Solution for Traction Motor Used in Hybrid Vehicles. *4th International Symposium on Applied Computational Intelligence and Informatics*. IEEE, 2007. Adapted from Figure 2, p. 2

## Cyber Exploration Laboratory

### Experiment 3.1

**Objectives** To learn to use MATLAB to (1) generate an LTI state-space representation of a system and (2) convert an LTI state-space representation of a system to an LTI transfer function.

**Minimum Required Software Packages** MATLAB and the Control System Toolbox

**Prelab**

1. Derive the state-space representation of the translational mechanical system shown in Skill-Assessment Exercise 3.2 if you have not already done so. Consider the output to be  $x_3(t)$ .
2. Derive the transfer function,  $\frac{X_3(s)}{F(s)}$ , from the equations of motion for the translational mechanical system shown in Skill-Assessment Exercise 3.2.

**Lab**

1. Use MATLAB to generate the LTI state-space representation derived in Prelab 1.
2. Use MATLAB to convert the LTI state-space representation found in Lab 1 to the LTI transfer function found in Prelab 2.

**Postlab**

1. Compare your transfer functions as found from Prelab 2 and Lab 2.
2. Discuss the use of MATLAB to create LTI state-space representations and the use of MATLAB to convert these representations to transfer functions.

**Experiment 3.2**

**Objectives** To learn to use MATLAB and the Symbolic Math Toolbox to (1) find a symbolic transfer function from the state-space representation and (2) find a state-space representation from the equations of motion.

**Minimum Required Software Packages** MATLAB, the Symbolic Math Toolbox, and the Control System Toolbox

**Prelab**

1. Perform Prelab 1 and Prelab 2 of Experiment 3.1 if you have not already done so.
2. Using the equation  $T(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}$  to find a transfer function from a state-space representation, write a MATLAB program using the Symbolic Math Toolbox to find the symbolic transfer function from the state-space representation of the translational mechanical system shown in Skill-Assessment Exercise 3.2 and found as a step in Prelab 1.
3. Using the equations of motion of the translational mechanical system shown in Skill-Assessment Exercise 3.2 and found in Prelab 1, write a symbolic MATLAB program to find the transfer function,  $\frac{X_3(s)}{F(s)}$ , for this system.

**Lab**

1. Run the programs composed in Prelabs 2 and 3 and obtain the symbolic transfer functions by the two methods.

**Postlab**

1. Compare the symbolic transfer function obtained from  $T(s) = \mathbf{C}(s\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}$  with the symbolic transfer function obtained from the equations of motion.
2. Discuss the advantages and disadvantages between the two methods.
3. Describe how you would obtain an LTI state-space representation and an LTI transfer function from your symbolic transfer function.

**Experiment 3.3**

**Objectives** To learn to use LabVIEW to (1) generate state-space representations of transfer functions, (2) generate transfer functions from state-space representations, and (3) verify that there are multiple state-space representations for a transfer function.



**Minimum Required Software Packages** LabVIEW, the LabVIEW Control Design and Simulation Module, and the MathScript RT Module.

### Prelab

1. Study Appendix D, Sections D.1 through D.4, Example D.1.
2. Solve Skill-Assessment Exercise 3.3.
3. Use your solution to Prelab 2 and convert back to the transfer function.

### Lab

1. Use LabVIEW to convert the transfer function,  $G(s) = \frac{2s + 1}{s^2 + 7s + 9}$ , into a state-space representation using both the graphical and MathScript approaches. The front panel will contain controls for the entry of the transfer function and indicators of the transfer function and the two state-space results. Functions for this experiment can be found in the following palettes: (1) **Control Design and Simulation/Control Design/Model Construction**, (2) **Control Design and Simulation/Control Design/Model Conversion**, and (3) **Programming/Structures** Hint: Coefficients are entered in reverse order when using MathScript with MATLAB.
2. Use LabVIEW to convert all state-space representations found in Lab 1 to a transfer function. All state-space conversions should yield the transfer function given in Lab 1. The front panel will contain controls for entering state-space representations and indicators of the transfer function results as well as the state equations used.

### Postlab

1. Describe any correlation found between the results of Lab 1 and calculations made in the Prelab.
2. Describe and account for any differences between the results of Lab 1 and calculations made in the Prelab.
3. Explain the results of Lab 2 and draw conclusions from the results.

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